

# SAE Aero Design

### Group 10

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#### CELEBRATING 25 YEARS!

# Motivation **AERO DESIGN**

#### SAE Collegiate Design Series

# Project Specifications (Rules)

### Aircraft Dimension Requirement

 Fully configured for takeoff, the free standing aircraft shall have a maximum combined length, width, and height of 225 inches. Aircraft exceeding this design requirement will be disqualified from the competition.

### Gross Weight Limit

 Regular Class aircraft may not weigh more than fifty five (55) pounds with payload and fuel.

#### Engine Requirements

 Regular Class aircraft can still be powered by a single, unmodified O.S 61FX with

#### Standard Design





#### "Flying Wing"





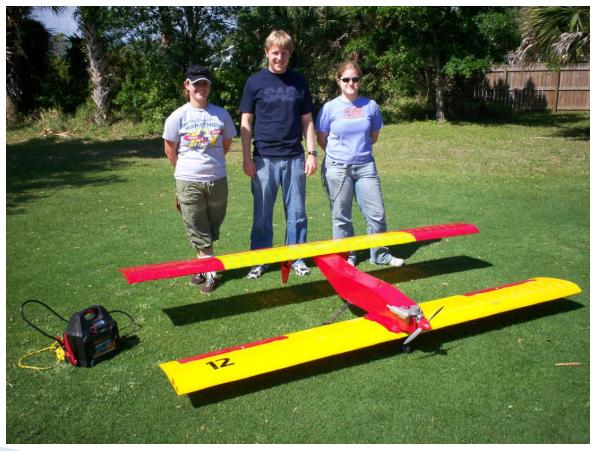
NASA Dryden Flight Research Center Photo Collection http://www.dfrc.nasa.gov/gallery/photo/index.html NASA Photo: ED01-0209-3 Date: July 14, 2001 Photo by: Nick Galante/PMRF The Helios Prototype flying wing is shown over the Pacific Ocean during its first test flight on solar power from the U.S. Navy's Pacific Missile Range Facility in Hawaii.

NASA

#### Minimalist Design



#### Double Wing



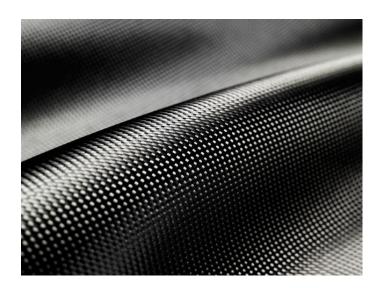
#### Biplane





## Materials

- Carbon fiber
- Advanced composites
- Balsa wood/heat shrink skin







## **Decision Matrix**

#### **Decision Matrix**

Rating: 1 - 10		Concepts										
		Standard Design		"Flying Wing" Design		Minimalist Design		Canard Wing Design		Bi-Plane Design		
Selection Criteria	Weight	Rating	Weighed Score	Rating	Weighed Score	Rating	Weighed Score	Rating	Weighed Score	Rating	Weighed Score	
Potential Lift	20%	7	1.4	9	1.8	8	1.6	8	1.6	7	1.4	
Potential Drag	10%	4	0.4	8	0.8	9	0.9	2	0.2	3	0.3	
Durability	15%	9	1.35	5	0.75	3	0.45	7	1.05	7	1.05	
Cost	10%	5	0.5	5	0.5	8	0.8	3	0.3	4	0.4	
Ease of Build	5%	5	0.25	6	0.3	8	0.4	4	0.2	4	0.2	
Potential Flight Score	e 40%	8	3.2	6	2.4	7	2.8	7	2.8	7	2.8	
	Total Score	7.1		6.55		6.95		6.15		6.15		
	Rank	1		3		2		4(tied)		4(tied)		

### **Conclusion and Future Plans**

- More analysis to see if our decision matrix results are worthy
- Re-examine design opportunities and finalize on one type of design
- Once one whole design is chosen, examine design possibilities for each component

Reference(s)							
http://students.sae.org/competitions/aerodesign/east/							
http://www.uwindsor.ca/aero/system/files/plane.jpg http://www.airliners.net/aviation- forums/non_aviation/print.main?id=1323590							
http://www.af.mil/shared/media/photodb/photos/030922-F-0000J- 888.jpg							
http://aerodesign.meil.pw.edu.pl/ad_new/wp- content/uploads/2011/03/IMG_6904na_strone.jpg							
http://my.fit.edu/cargoplane/img/taxi1%20011.jpg							
http://www.engin.umich.edu/teamprojects/teams/aero/teamprojects/te ams/aero/mfly4.jpg http://iignite.com/SAE/Gallery/content/bin/images/large/_0114022654. jpg							
http://dsgperformance.files.wordpress.com/2010/03/carbon-fiber- frame-lg.jpg http://www.airfieldmodels.com/gallery_of_models/rc/wermachts_scorpi on/images/17581.jpg							

#### ANY QUESTIONS?