

## Virtual Design Review 2

Team 507 - SAE Aero Design – Aero and Propulsion Team



## **Team Introductions**

Sasindu Pinto: Project /Aeronautics/Propulsion Engineer Noah Wright: Aerodynamics Engineer



Cameron Riley: Materials/Hardware Engineer



Michenell Louis-Charles: Thermal Fluids Engineer/Financial Chair



Adrian Moya: Systems/Hardware Engineer





## **Sponsor and Advisors**



Florida Space Grant Consortium: Funding Sponsor Seminole RC Club: Equipment/Personnel Sponsor

Dr. Chiang Shih: Professor & AME Center Director Advisor

Presenter: CR



## **Team Objective**



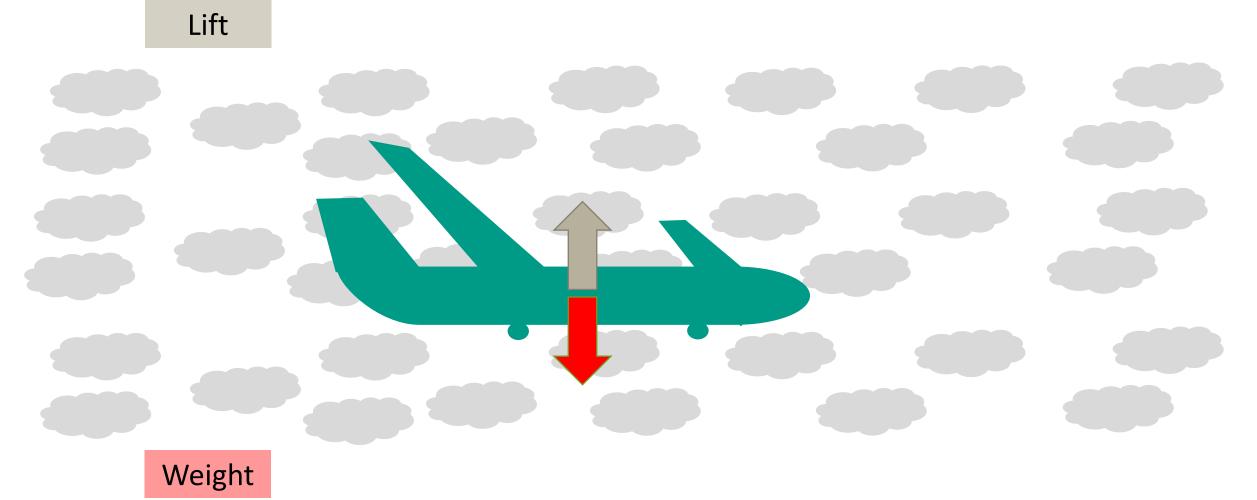
The objective of the aero-propulsion team is to ensure that the plane takes off and lands while carrying a payload while completing the flight path.

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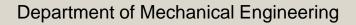


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## **Key Definitions**

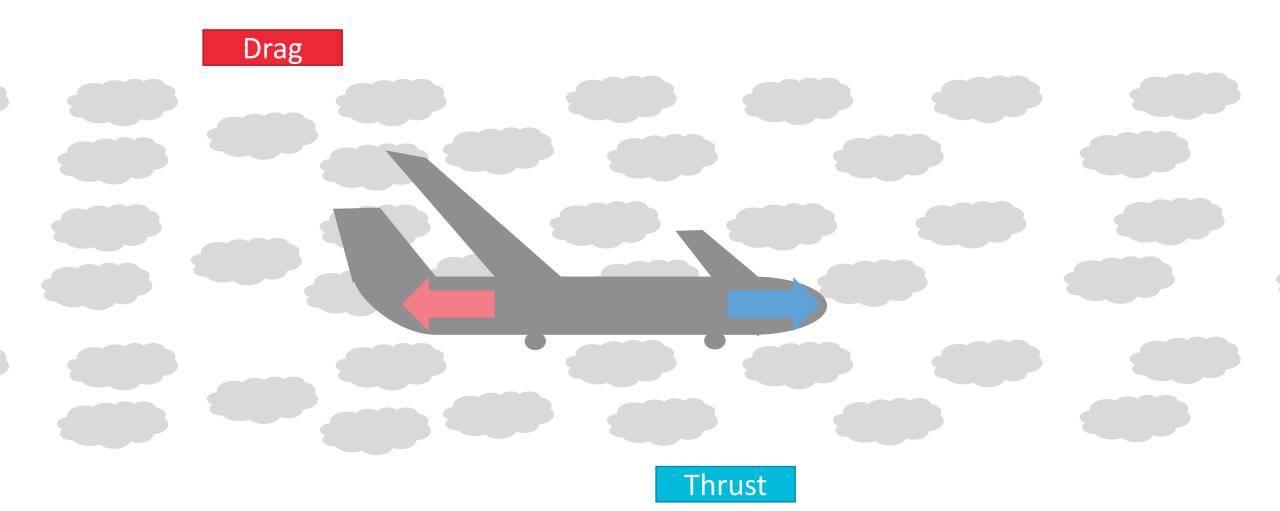


Presenter: CR





## Key Definitions



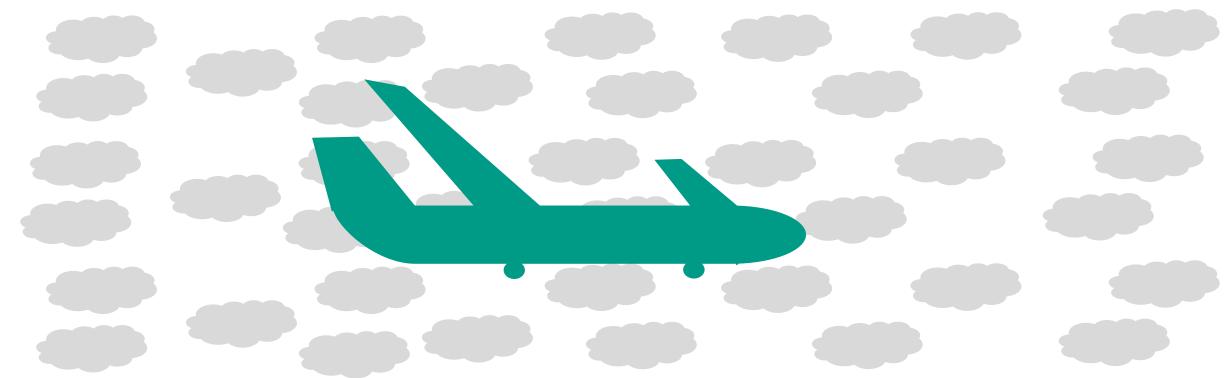
Presenter: CR







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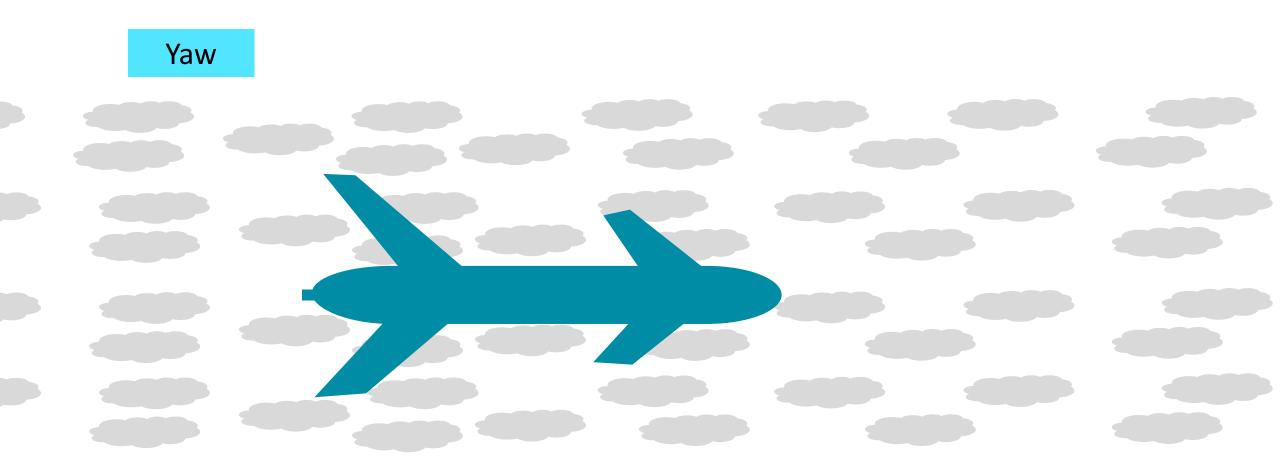


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# **VDR 1 Review**

**Presenter – Cameron Riley** 



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## **Project Background**



- Plane to be entered in SAE Aero Design Competition East

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## Key Goals



- The plane takeoff, cruise, and land while carrying a cargo load
- The plane carries a minimum of one soccer ball as the cargo load



- Will be flown in atmospheric conditions at sea level
- Motors and electronics will be store bought and not custommade



Presenter: CR



## Markets and Stakeholders



Markets

- m Primary
  - **SAE** Aero Design Competition
  - Madvisors
  - Sponsors
- m Secondary
  - m Professionals in the Aviation field
  - **m** Aviation Companies
  - RC Hobbyists
  - Scholars that reference this project



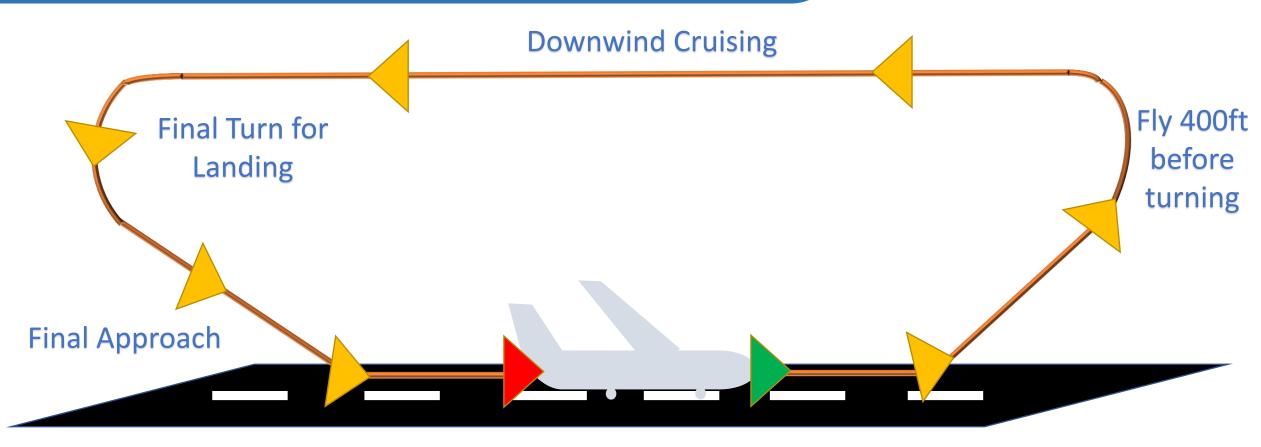
- Dr. McConomy and Dr. Shih
- FAMU-FSU College of Engineering
- SAE Design Competition
- 🕾 RC Pilots
- Senior Design Teams 507&508

Presenter: CR





## **Customer Needs**



### Land within 400 ft

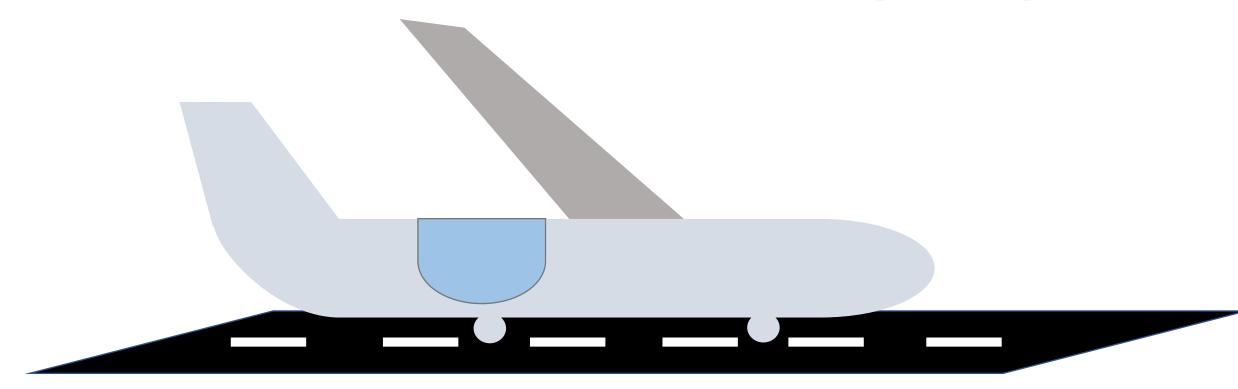
### Takeoff within 100 ft

Presenter: CR



## **Customer Needs**

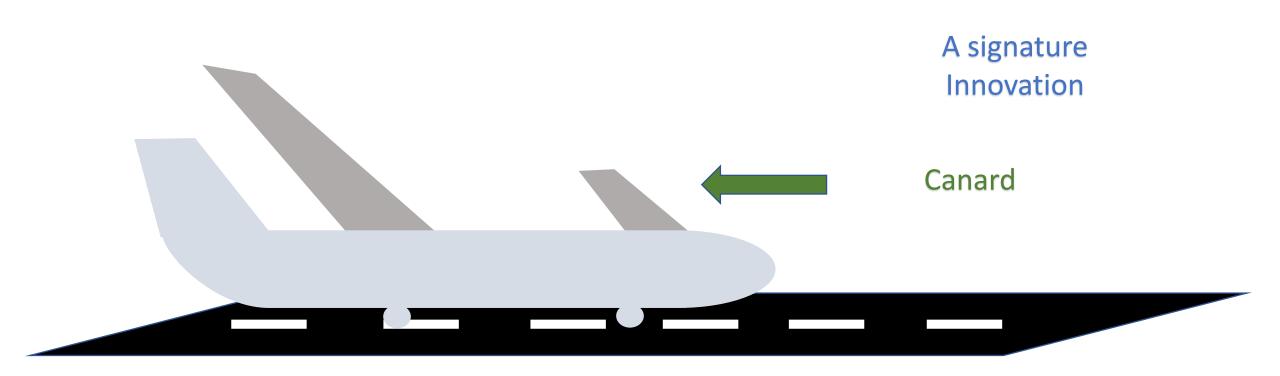
#### Loading/Unloading time – 1 min



Presenter: CR



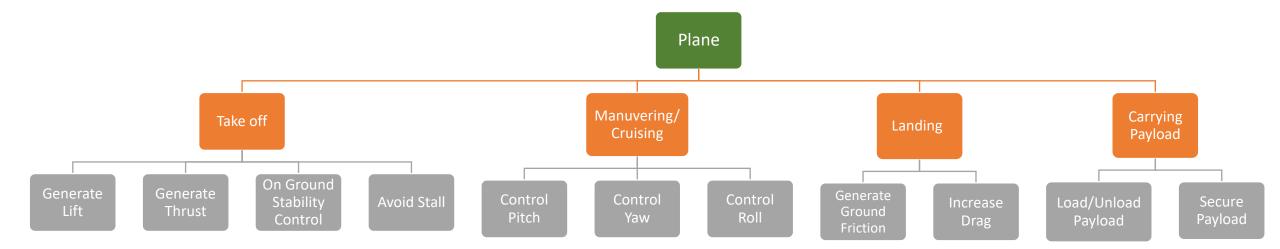
## **Customer Needs**



Presenter: CR



## **Functional Decomposition**



Presenter: CR



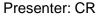


**Presenter – Cameron Riley** 



- Kienenagtelidfrastack (AoA)
  - Cheffic Forder With Diagran tainfoil
  - Eternidiern Digrodeter ~ 1693-294m 1
- Acteterationean Aerodynamic
  - **ጥ**ብ የ መጀን የ መጀ
- Stall Speed
  - 25 mph







### Control Surface Movement

- Controlling pitch ~ angle about x-axis
  - Servo Motor Torque produced ~ Greater than 66 oz-in
- Controlling yaw ~ angle about y axis
  - Servo Motor Torque produced ~ Greater than 66 oz-in
- Controlling roll ~ angle about z axis
  - Servo Motor Torque produced ~ Greater than 66 oz-in
- Weight
  - Less than 55 lbs



### • Generating Drag (Air and Ground)

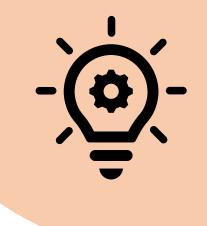
- Coefficient of Drag ~ Greater than 1
- Landing Velocity ~ 25 mph
- Deceleration
  - Air Brake Force ~ 2-5 lbf
  - Coefficient of Rolling Friction ~ 0.03-0.06
  - Landing Distance 400 ft











# Concept Generation

Presenter – Adrian Moya

Department of Mechanical Engineering



## **Concept Generation**

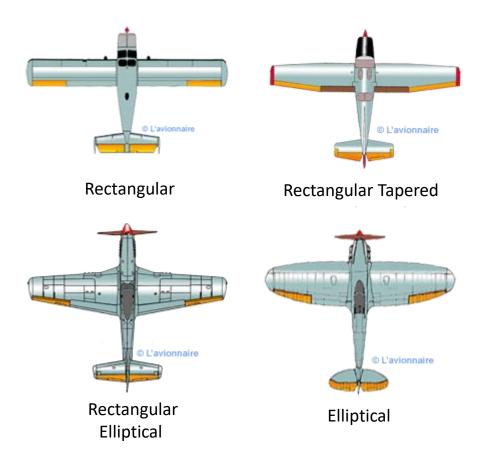
- Methods used
  - Morphological Analysis
  - Biomimicry
  - Competitive Benchmarking
  - Crapshoot

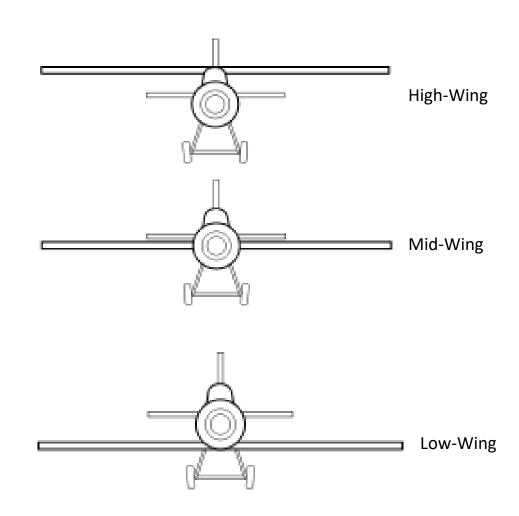


Presenter: AM



## **Design concepts**



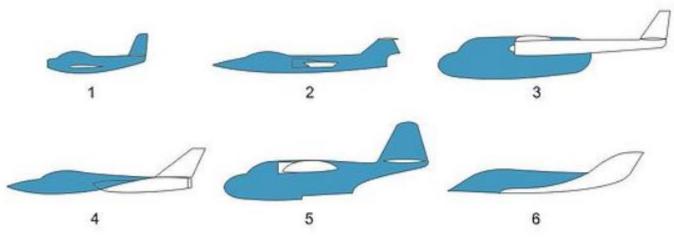




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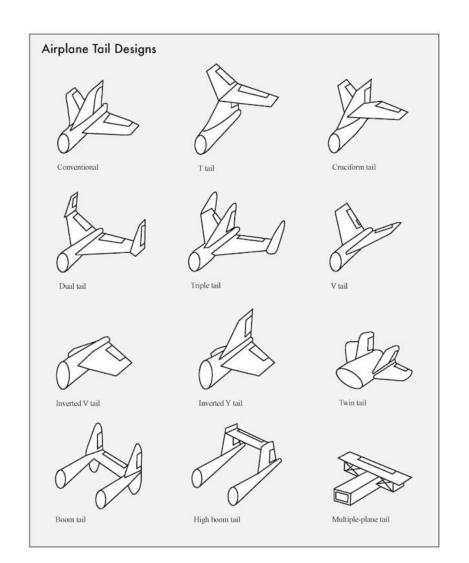
## **Design Concepts**

### Fuselage Layouts



- 1: <u>Subsonic</u>
- 2: High-speed / supersonic
- 3: High-capacity subsonic

- 4: High-maneuverability supersonic
- 5: <u>Flying boat</u>
- 6: <u>Hypersonic</u>





## Morphological Analysis

#### Morphological Analysis Wing Control Wing Layout Wing Type Fuselage Tail Position Surfaces Main - Forward Swept **Boom-Mounted** Main-Tail High Wing Aileron Bullet Tail- Symmetric (x-29) Inverted V Flying Boat Trapezoidal Delta Wing Mid Wing Flaps H-Tail Main - Elliptical Tail -Twin-Tail Canard-Main Low Wing Elevators Double Boom Symmetric Main - Trapezoidal Tail **Tapered** - Symmetric Triple-Tail Y-Tail

## **Competitive Benchmarking**

Rutan Long E-Z: Small composite plane with canards & tip sails



Cessna 208 Grand Caravan: Typical bush plane with extra cargo space



Kawasaki C-2: Japanese military cargo plane





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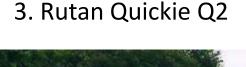
## **Medium and High Fidelity Concepts**

1. Boomtown



2. Rutan Long EZ





4. Boeing 747 Dreamlifter



DREAM)LIFTER

5. Cessna 208 Grand Caravan

6. OMAC Laser 300

7. Aero Spacelines Super Guppy







8. Kawasaki C-2



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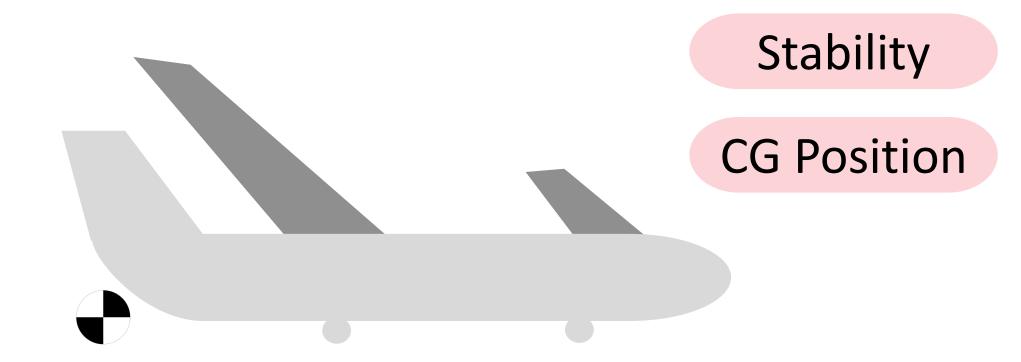


## **Concept Selection**

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## Takeoff/Landing Requirements

Land within 400 ft

#### Takeoff within 100 ft

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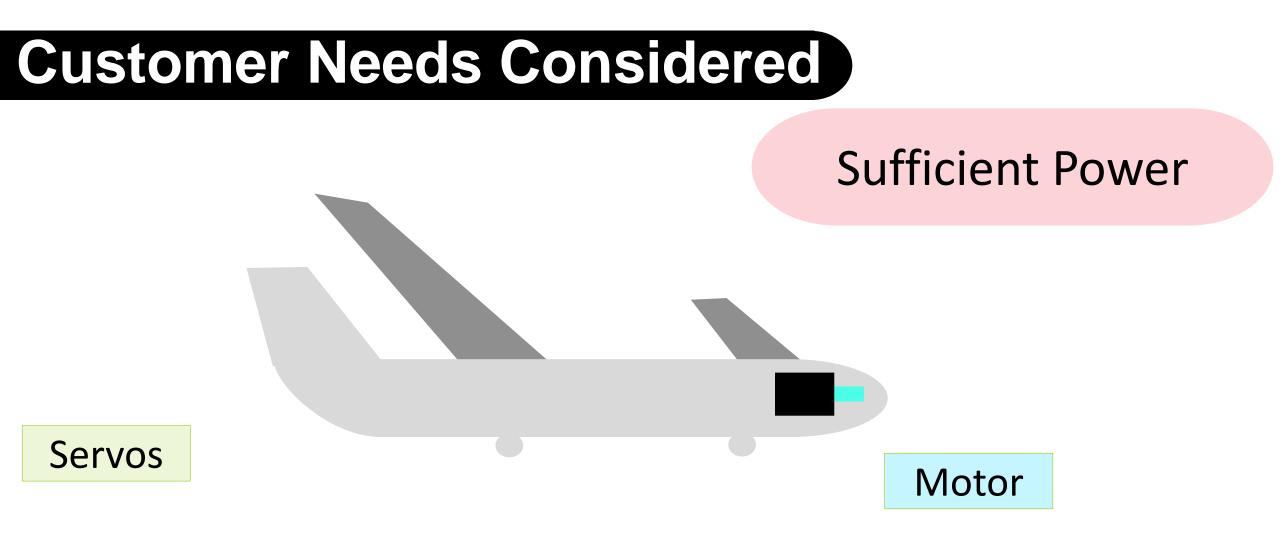
## Wingspan Requirements

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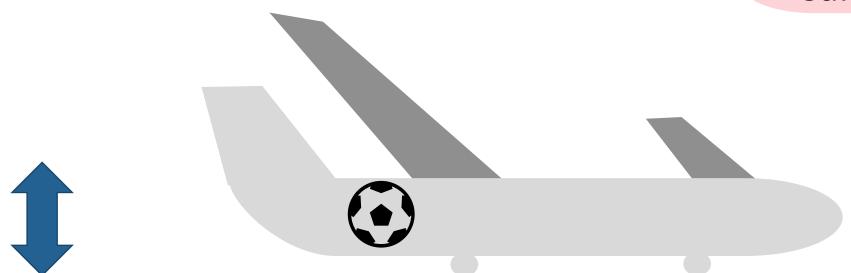


## **Ground Controls Front Wheel** Control Rudder **Elevators/Ailerons**

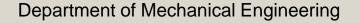
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### **Carrying Cargo Load**



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## **Binary Pairwise Comparison**

Binary Pairwise Comparison													
	1	2	3	4	5	6	7	8	9	10	11	12Total	
1. Material	-	0	0	0	0	0	0	1	0	0	0	0	1
2. Stability	1	-	0	0	0	1	1	1	1	0	0	1	6
3. CG in front of CP	1	1	-	1	1	1	1	1	1	1	1	1	10
			0				1	0	1	0	0		-
4. Meet takeoff/landing requirements	1	1	0	-	1	1	1	0	1	0	0	T	/
5. Wingspan meets restrictions	1	1	0	0	-	1	1	1	1	0	0	1	7
6. Sufficient Power	1	0	0	0	0	-	0	0	1	1	1	1	5
7. Maneuverability	1	0	0	0	0	1	-	0	1	0	0	1	4
8. Light Weight	0	0	0	1	0	1	1	-	1	1	0	1	6
9. Touch-down Impact	1	0	0	0	0	0	0	0	-	0	0	1	2
10. Ground Controls	1	1	0	1	1	0	1	0	1	-	1	1	7
11. Carry the Minimum Cargo Load													
Required	1	1	0	1	1	0	1	1	1	0	-	1	8
12. Easy to Load/Unload	1	0	0	0	0	0	0	0	0	0	0	-	1
Total	10	5	0	4	4	6	7	5	9	4	3	10	-

Presenter: AM



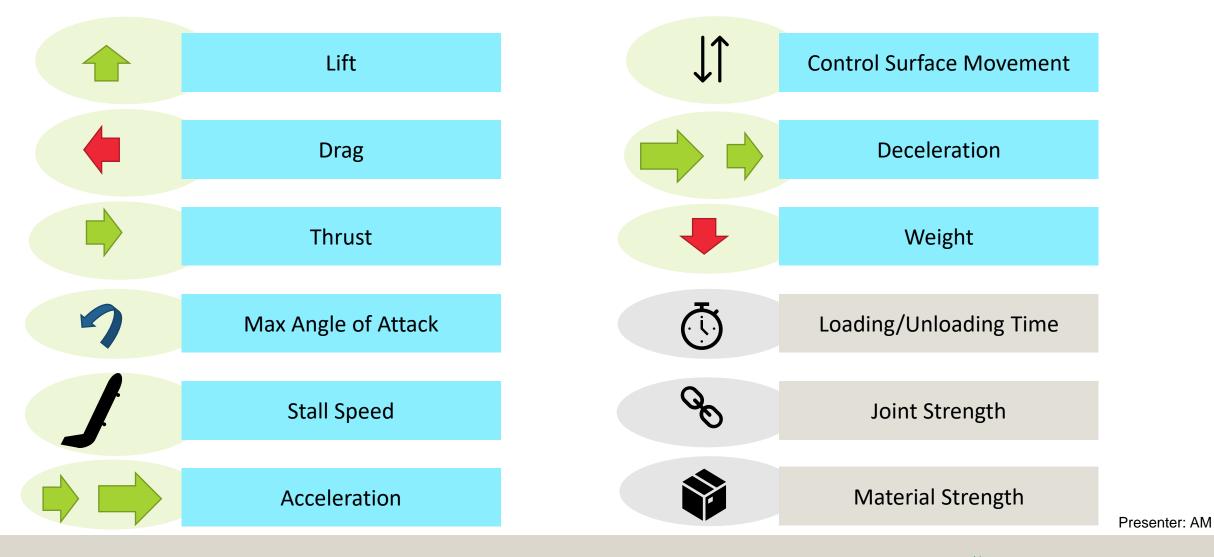
### **Binary Pairwise Comparison**

	Binar	y Paiı	rwise	Comp	bariso	n							
	1	2	3	4	5	6	7	8	9	10	11	121	otal
1. Material	-	0	0	0	0	0	0	1	0	0	0	0	1
2. Stability	1	-	0	0	0	1	1	1	1	0	0	1	6
3. CG in front of CP	1	1	-	1	1	1	1	1	1	1	1	1	10
<ol> <li>Meet takeoff/landing requirements</li> </ol>	1	1	0	_	1	1	1	0	1	0	0	1	7
	-	-	Ŭ		-	-	-	U	-	U	Ŭ	-	,
5. Wingspan meets restrictions	1	1	0	0	-	1	1	1	1	0	0	1	7
6. Sufficient Power	1	0	0	0	0	-	0	0	1	1	1	1	5
7. Maneuverability	1	0	0	0	0	1	-	0	1	0	0	1	4
8. Light Weight	0	0	0	1	0	1	1	-	1	1	0	1	6
9. Touch-down Impact	1	0	0	0	0	0	0	0	-	0	0	1	2
10. Ground Controls	1	1	0	1	1	0	1	0	1	-	1	1	7
11. Carry the Minimum Cargo Load													
Required	1	1	0	1	1	0	1	1	1	0	-	1	8
12. Easy to Load/Unload	1	0	0	0	0	0	0	0	0	0	0	-	1
Total	10	5	0	4	4	6	7	5	9	4	3	10	-

Presenter: AM



#### **Engineering Characteristics**







#### Concept Selection House Of Quality

Presenter – Sasindu Pinto

Department of Mechanical Engineering



					House of Qu								
					Engine	ering Ch	aracterist	ics (***From I	Main Targets*	**)			
Improvement Direction		1	➡	1	<b>1</b>	1	1	1	➡	➡	<b>1</b>		
Units	I	bf Il	of Ik	of o	degrees	ft/s	ft/s^2	degrees	seconds	lbs	ft/s^2	psi p	osi
Customer Requirements 1. Material	Importance Weight Factor	Lift	Drag 1	Thrust	Max Angle of Attack	Stall Speed	Acceleration	Control Surface Movement	Loading/ Unloading Time	weight		م Joint Strength	Material م Strength
2. Stability	6	9	3	3				g					
3. CG in front of CP	10	9	3	9	9	g		9		3			
<ol> <li>Meet takeoff/landing requirements</li> </ol>	7	9	3	9			ç	)			9		
5. Wingspan meets restrictions	7	9	3		3	3		1				3	3
6. Sufficient Power	5	1	1	3			3			1	_		
7. Maneuverability	4		_	_	3	3		S		3		3	1
8. Light Weight	6	3		3			3			9			
9. Touch-down Impact	2							3		3	9	9	9
10. Ground Controls	7							1					
11. Carry the Minimum Cargo Load Required	8	9		3			3	3	g	9	3	9	9
12. Easy to Load/Unload	1								9	3		3	
Raw Score		365	96	228	123		120	) 215			128	135	124
Relative Weight %		18.92	4.98	11.82	6.38	6.38	6.22	2 11.15			6.64	7.00	6.43
Rank Order		1	11	2	6	6	5 10	) 3	12	. 4	. 8	5	9



					House of Qu	-							
		•		•	Engine	ering Ch	aracterist	ics (***From N	Main Targets*	**)	•	•	
Improvement Direction		1	➡	1	1	1	1	1	•	➡	1		
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Customer Requirements	Importance Weight Factor	Lift	Drag	Thrust	Max Angle of Attack	Stall Speed	Acceleration	Control Surface Movement	Loading/ Unloading Time	-			Material Strength
1. Material	1	0	1	2						9		9	9
2. Stability 3. CG in front of CP	6 10	9 9	3	3 9	g	ç		9		2			
	10	9	3	9	5	5		9		3			
<ol> <li>Meet takeoff/landing requirements</li> </ol>	7	9	3	9			ç	)			9		
5. Wingspan meets restrictions	7	9	3		3	3		1				3	3
6. Sufficient Power	5	1	1	3			3			1	. 1		
7. Maneuverability	4				3	3		9		3		3	1
8. Light Weight	6	3		3			3			9		-	0
9. Touch-down Impact	2							3		3	9	9	9
10. Ground Controls 11. Carry the Minimum Cargo Load	/							1					
Required	8	9		3			Э	8	9	9	3	9	9
12. Easy to Load/Unload	1								9	3		3	
Raw Score		365	96	228	123	123	120	) 215	81	191	128	135	124
Relative Weight %		18.92	4.98	11.82	6.38	6.38	6.22	11.15	4.20	9.90	6.64	7.00	6.43
Rank Order		1	11	2	6	6	5 10	) 3	12	. 4	. 8	5	9



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			_		Engine	ering Ch	aracterist	ics (***From N	viain Targets*	**)			_
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Customer Requirements 1. Material	Importance Weight Factor	Lift	Drag 1	Thrust	Max Angle of Attack	Stall Speed	Acceleration	Control Surface Movement	Loading/ Unloading Time	o Weight		o Joint Strength	Material م Strength
2. Stability	6	9	3	3				9					<u> </u>
3. CG in front of CP	10	9	3	9		9		9		3			
4. Meet takeoff/landing requirements	7	9	3	9			9				ç	9	
5. Wingspan meets restrictions	7	9	3		3	3		1				3	3
6. Sufficient Power	5	1	1	3			3			1			
7. Maneuverability	4				3	3		9		3		3	1
8. Light Weight	6	3		3			3			9			
9. Touch-down Impact	2							3		3	<u>c</u>	9 9	9
10. Ground Controls	7							1					
11. Carry the Minimum Cargo Load Required	8	9		3			3		ç	9 9		3 9	9
12. Easy to Load/Unload	1								ç	) 3		3	
Raw Score		365	96	228	123	123	120	215	81	191	. 128	3 135	124
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Rank Order		1	11	2	E	6	10	3	12	2 4		3 5	9



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Customer Requirements	Importance Weight Factor	Lift	Drag	Thrust	Max Angle of Attack	Stall Speed	Acceleration	Control Surface Movement	Loading/ Unloading Time	. Weight	Deceleration	· · · ·	Material Strength
1. Material	1	-	1	-				-		9		9	9
2. Stability	6	9	3	3		0		9		-			
3. CG in front of CP	10	9	3	9	9	9		9		3			
<ol> <li>Meet takeoff/landing requirements</li> </ol>	7	9	3	9			9				ç		
5. Wingspan meets restrictions	7	9	3		3	3		1				3	3
6. Sufficient Power	5	1	1	3			3	3		1	. 1	-	
7. Maneuverability	4				3	3		9		3		3	1
8. Light Weight	6	3		3			3			9			
9. Touch-down Impact	2							3		3	ç	9	9
10. Ground Controls	7							1					
11. Carry the Minimum Cargo Load Required	8	9		3			3		g	9 9	3	9	9
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					Engine	ering Ch	aracterist	tics (***From I	Main Targets*	**)			
Improvement Direction		1	➡	1	1	1	1	1	•	₽	1		
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Customer Requirements 1. Material	Importance Weight Factor	Lift	Drag	Thrust	Max Angle of Attack	Stall Speed	Acceleration	Control Surface Movement	Loading/ Unloading Time	weight		to Joint Strength	
2. Stability	6	9	3	3				g					
3. CG in front of CP	10	9	3	9		) 9		g		3	3		
4. Meet takeoff/landing requirements	7	9	3	9				9			g	)	
5. Wingspan meets restrictions	7	9	3		3	3		1				3	3
6. Sufficient Power	5	1	1	3			3	3 3		1	. 1		
7. Maneuverability	4				3	3 3	8	9		Э	8	3	1
8. Light Weight	6	3		3			3	-		ç			
9. Touch-down Impact	2							3		3	s c	9 9	9
10. Ground Controls	7							1					
11. Carry the Minimum Cargo Load Required	8	9		3			3	3	ç	) <u>c</u>	) 3	9	9
12. Easy to Load/Unload	1								ç	) 3	3	3	
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				•	Engine	ering Ch	aracterist	tics (***From I	Main Targets*	**)			
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Units	I	bf II	bf Ik	of	degrees	ft/s	ft/s^2	degrees	seconds	lbs	ft/s^2	psi	psi
Customer Requirements	Importance Weight Factor	Lift	Drag	Thrust	Max Angle of Attack	Stall Sneed	Acceleration	Control Surface Movement	Loading/ Unloading Time	Weight		,	
1. Material	1		1							ç		9	9
2. Stability	6	9	3	3				S					
3. CG in front of CP	10	9	3	9		9 9	Ð	ç		3	6		
<ol> <li>Meet takeoff/landing requirements</li> </ol>	7	9	3	9			0	Э			ç	)	
5. Wingspan meets restrictions	7	9	3		:	3 3	3	1				3	3
6. Sufficient Power	5	1	1	3			3	3 3		1	. 1		
7. Maneuverability	4				:	3	3	9		3	8	3	1
8. Light Weight	6	3		3			3	3		ç	) 3		
9. Touch-down Impact	2							3		Э	9	) 9	9
10. Ground Controls	7							1					
11. Carry the Minimum Cargo Load Required	8	9		3			3	3	9	) <u>c</u>	) 3	9	9
12. Easy to Load/Unload	1								9	) 3		3	
Raw Score		365	96	228	12	3 -123	3 — 12	215	81	. 191	. 128	135	124
Relative Weight %		18.92	4.98	11.82	6.3	8 38	3 6.2.	11.15	4.20	9.90	6.64	7.00	6.43
Rank Order		1	11	2				З	12	2 4	s ا	5 5	9

Presenter: SP

Department of Mechanical Engineering



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			_		Engine	ering Ch	aracterist	ics (***From N	Main Targets*	**)			
Improvement Direction		T	➡	T	Ť	1	T	Ť	+	➡	T	Î	
Units	1	bf I	bf l	bf	degrees	ft/s	ft/s^2	degrees			ft/s^2	psi p	osi
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2. Stability	6	9	3	3				9					5
3. CG in front of CP	10	9	3	9	ç	ç	)	9		3			
4. Meet takeoff/landing requirements	7	9	3	9			ç				9		
5. Wingspan meets restrictions	7	9	3		3	3	3	1				3	3
6. Sufficient Power	5	1	1	3			3	-		1	. 1		
7. Maneuverability	4				3	3	8	9		3		3	1
8. Light Weight	6	3		3			3			9			
9. Touch-down Impact	2							3		3	9	9	9
10. Ground Controls	7							1					
11. Carry the Minimum Cargo Load Required	8	9		3			3	8	9	9	3	9	9
12. Easy to Load/Unload	1								9	3		3	
Raw Score		365	96	228	123					191			124
Relative Weight %		18.92	4.98	11.82						9.90	6.64		6.43
Rank Order		1	11	2	e	έ	5 10	) 3	12	4	. 8	5	9



#### **Concepts Considered**





2. Rutan Long EZ



#### 3. Rutan Quickie Q2



4. Boeing 747 Dreamlifter



5. Cessna 208 Grand Caravan

6. OMAC Laser 300

7. Aero Spacelines Super Guppy







8. Kawasaki C-2



Presenter: SP





Pugh Chart 1		Concepts							
		ł	ligh			Me	ediu	ım	
Selection Criteria	2020 Competition Entry	1	2	3	4	5	6	7	8
Lift		+	+	+	-	-	+	-	-
Thrust		S	S	S	S	S	S	S	S
	DATUM								
Control Surface Movement	DATOW	+	+	+	+	S	+	S	S
Weight		-	S	-	-	-	S	-	S
Joint Strength		+	+	+	+	+	+	+	+
# of pluses		3	3	3	2	1	3	1	1
# of S's		1	2	1	1	2	2	2	3
# of Minuses		1	0	1	2	2	0	1	1

Presenter: SP



Pugh Chart 1		(	Concepts	
		High	Medium	
Selection Criteria	2020 Competition Entry		H	
Lift			and a	
Thrust		1		
Control Surface Movement Weight	DATUM			
Joint Strength			0	
# of pluses				
# of S's		<u>т</u> 2 .		
# of Minuses		1 0 2	22011	

Presenter: SP



Pugh Chart 1		Concepts							
		ł	ligh	)		Me	ediu	ım	
Selection Criteria	2020 Competition Entry	1	2	3	4	5	6	7	8
Lift		+	+	+	-	-	+	-	-
Thrust		S	S	S	S	S	S	S	S
	DATUM								
Control Surface Movement	DATOW	+	+	+	+	S	+	S	S
Weight		-	S	-	-	-	S	-	S
Joint Strength		+	+	+	+	+	+	+	+
# of pluses		3	3	3	2	1	3	1	1
# of S's		1	2	1	1	2	2	2	3
# of Minuses		1	0	1	2	2	0	1	1

Presenter: SP



Pugh Chart 2			Cor	ncept	5
			High	Me	edium
Selection Criteria	Concept 2		1	3	6
Lift		-	+	-	
Thrust		S	S	S	
	Datum				
Control Surface Movement	Datum	+	+	+	
Weight		-	-	-	
Joint Strength		S	S	S	
# of pluses			1	2	1
# of S's			2	2	2
# of Minuses			2	1	2

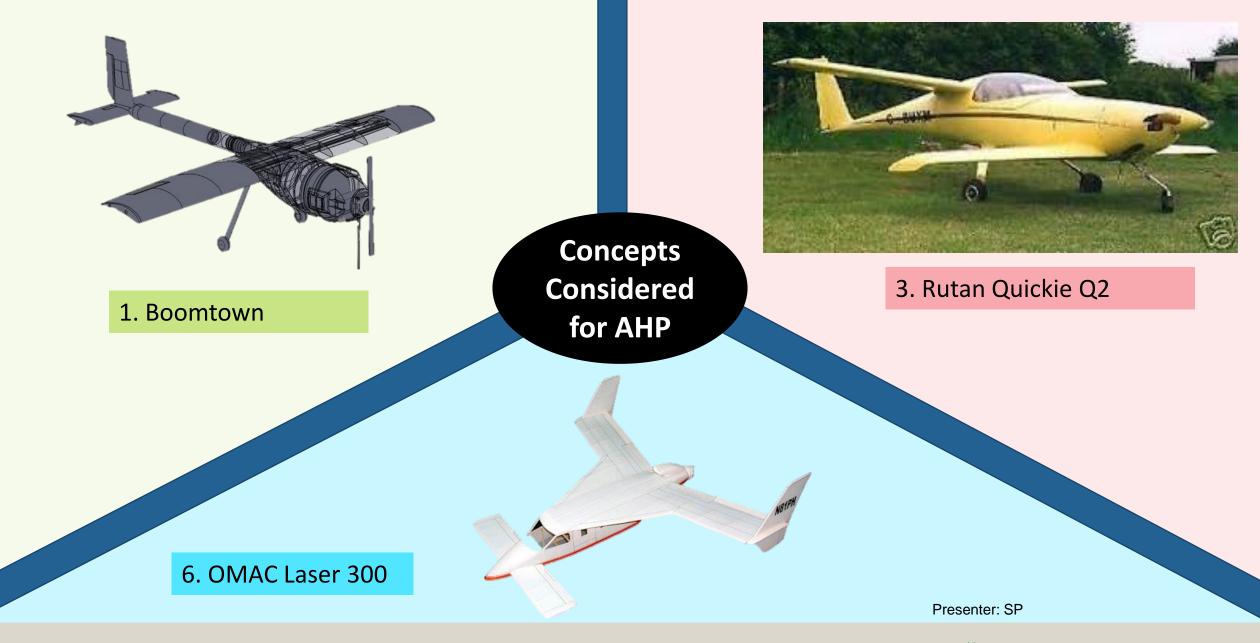


Pugh Chart 2	Concepts			S	
			High	Me	edium
Selection Criteria	Concept 2		1	3	6
Lift		-	+	-	
Thrust		S	S	S	
	Datum				
Control Surface Movement	Datum	+	+	+	
Weight		-	-	-	
Joint Strength		S	S	S	
# of pluses			1	2	1
# of S's			2	2	2
# of Minuses			2	1	2



Pugh Chart 2	Concepts				
		Hi	gh	Medi	um
Selection Criteria	Concept 2	1	-	3	6
Lift		-	+	-	
Thrust		S	S	S	
	Datum				
Control Surface Movement	Datum	+	+	+	
Weight		-	-	-	
Joint Strength		S	S	S	
# of pluses		1		2	1
# of S's		2		2	2
# of Minuses		2	) -	1	2

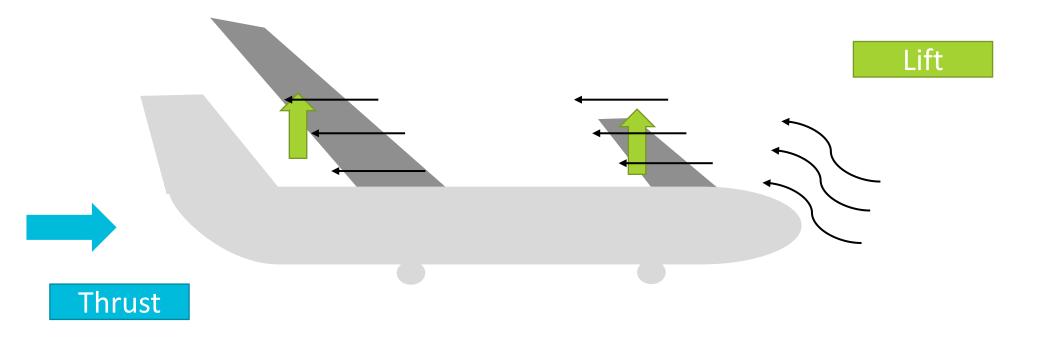






#### Criteria Comparison - AHP

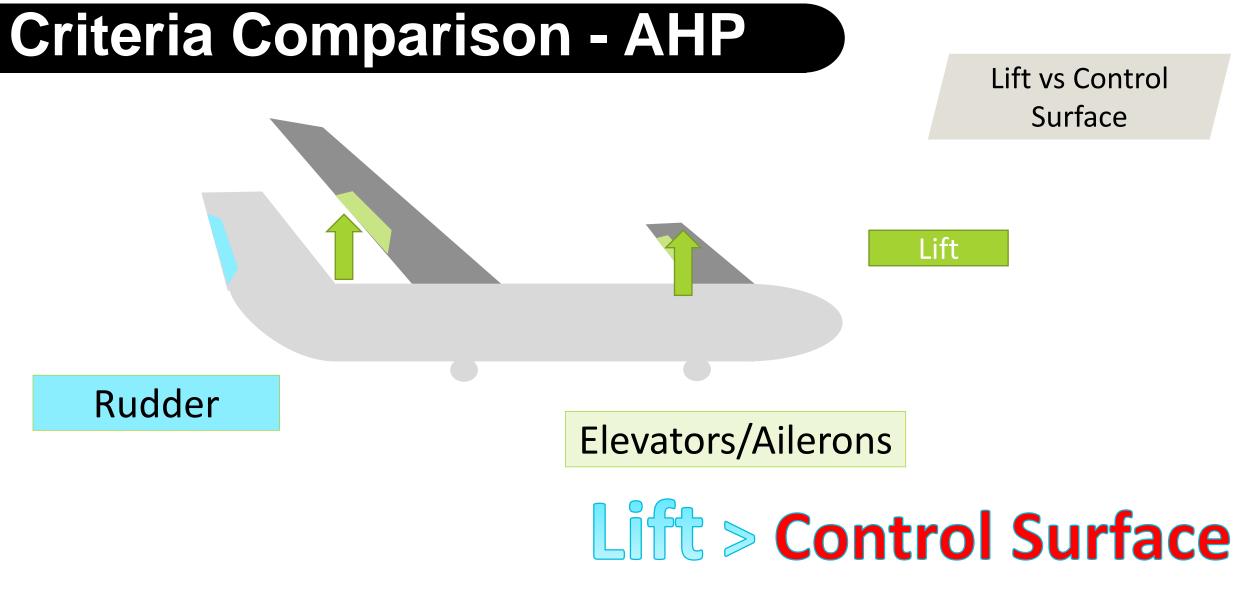
#### Lift vs Thrust



### Thrust > Lift

Presenter: SP



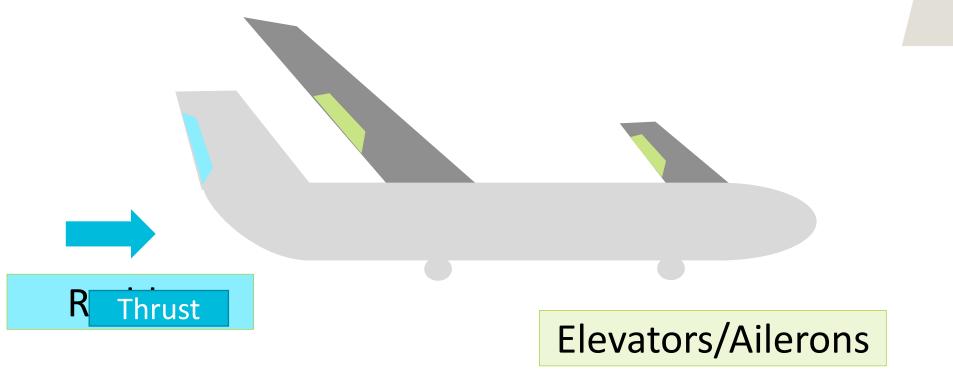


Presenter: SP



#### Criteria Comparison - AHP

Thrust vs Control Surface



### **Thrust > Control Surface**

Presenter: SP



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#### All Criteria Comparison Plots

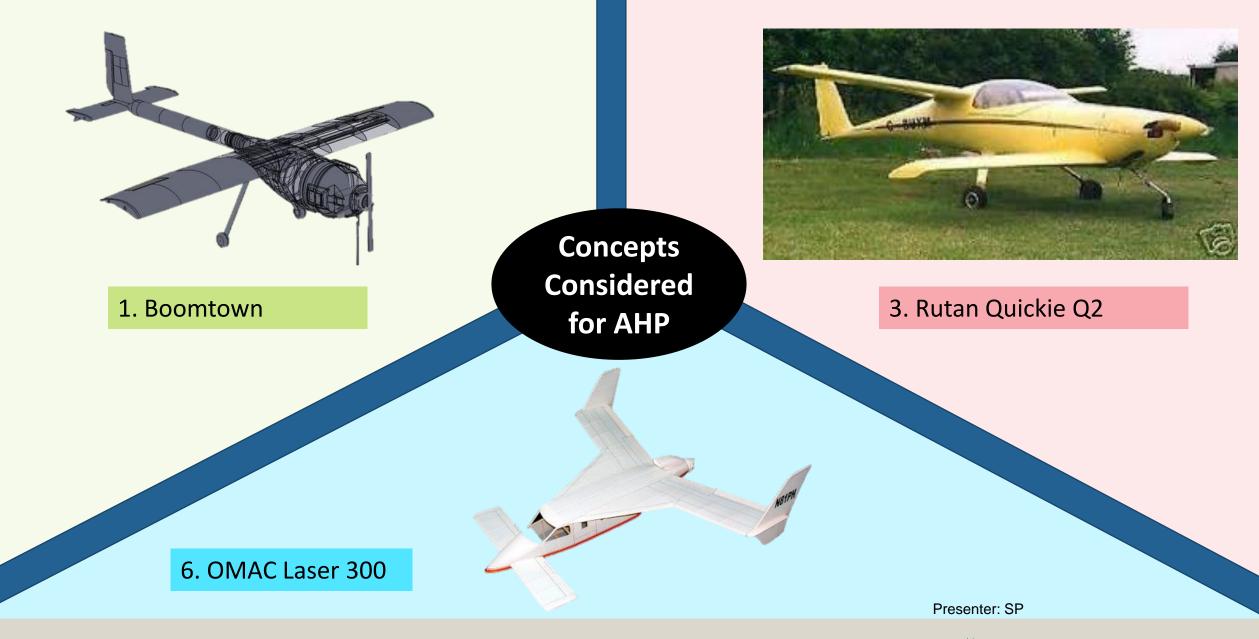
### **Criteria Comparison Matrix - AHP**

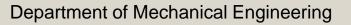
	Development	: of a Can	didate set of C	riteria W	eights {W}	
Criteria Comparison Matrix						
	Lift	Thrust	Control Surfac Movement	-	Weight	Joint Strength
Lift	1.00	0.33		3.00	9.00	9.00
Thrust	3.00	1.00		3.00	9.00	9.00
Control Surface Movement Weight Joint Strength	0.33 0.11 0.11	0.33 0.11 0.11		1.00 0.20 0.33	1.00	0.11
Sum	4.56	1.89		7.53	33.00	22.11
λ Average Consistency	CI Consistenc Index	y <mark>C</mark>	CR Consistency Ratio			CR<
6.053	3	0.027		0.051		

Presenter: SP

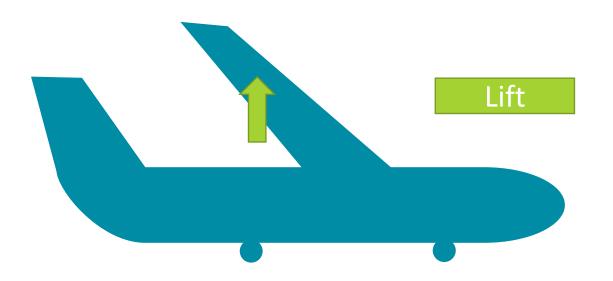


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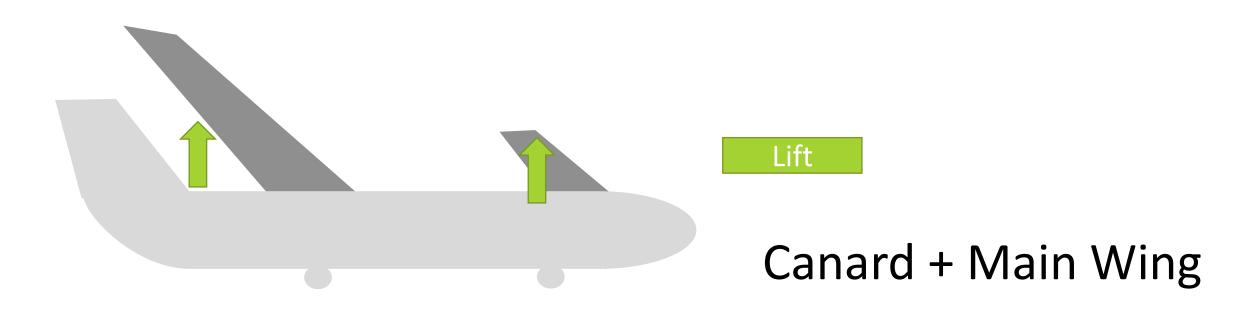


#### Just the main wing

#### Concept 1: Boomtown

Presenter: SP

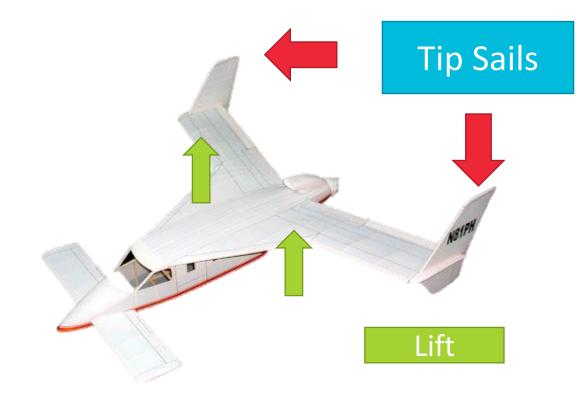




#### Concept 3: Rutan Quickie Q2

Presenter: SP



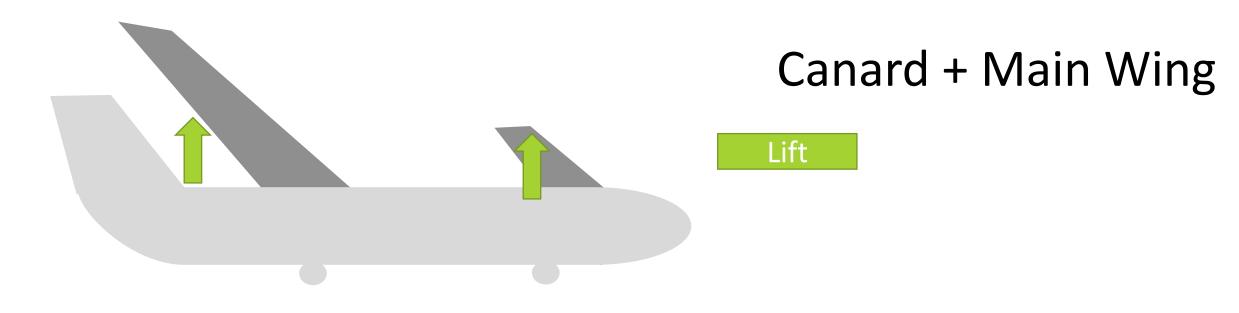


#### Lower Wingspan + Delta Restriction

Concept 6: OMAC 300 Laser Plane

#### Presenter: SP





## Concept 3 > Concept 6 > Concept 1

Concept 3: Rutan Quickie Q2

Concept 6: OMAC 300 Laser Plane

Concept 1: Boomtown

Presenter: SP





Lift Com	narie	on Ma	atrix - AH	Comparison for All	Criteria
	palis			Thrust CSM Weight	Joint Strength
	Lift Com	parison			★
	Concept	Concept			1
	1	3	Concept 6		
Concept 1	1.00	0.33	3.00		2
Concept 3	3.00	1.00	7.00		3
Concept 6	0.33	0.14	1.00		
Sum	4.33	1.48	11.00		
λ C Average C Consistency Ir 3.00703	Consistency <mark>Co</mark>		CR<0.1	Presenter: SP	6



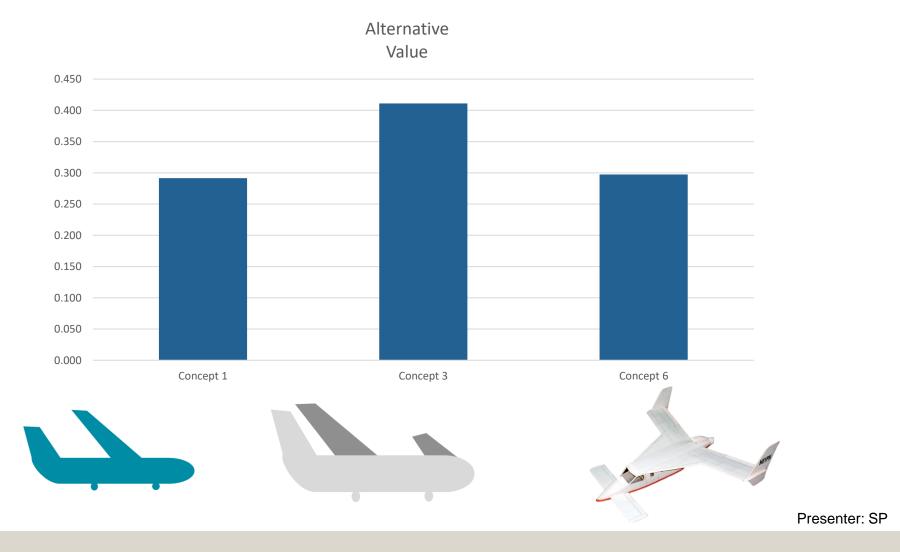
### Final Rating & Alternative Values - AHP



	Final Rati			Concept	Alternative Value			
Selection Criteria	Concept 1	Concept 2	Concept 6	concept	Vulue			
Lift	0.243	0.669	0.088	Concept 1	0.292			
Thrust	0.333	0.333	0.333	Concept 3	0.411			
Control Surface								
Movement	0.236	0.110	0.654					
Weight	0.260	0.633	0.106	Concept 6	0.297			
Joint Strength	0.333	0.333	0.333					
1 3 Presenter: SP								

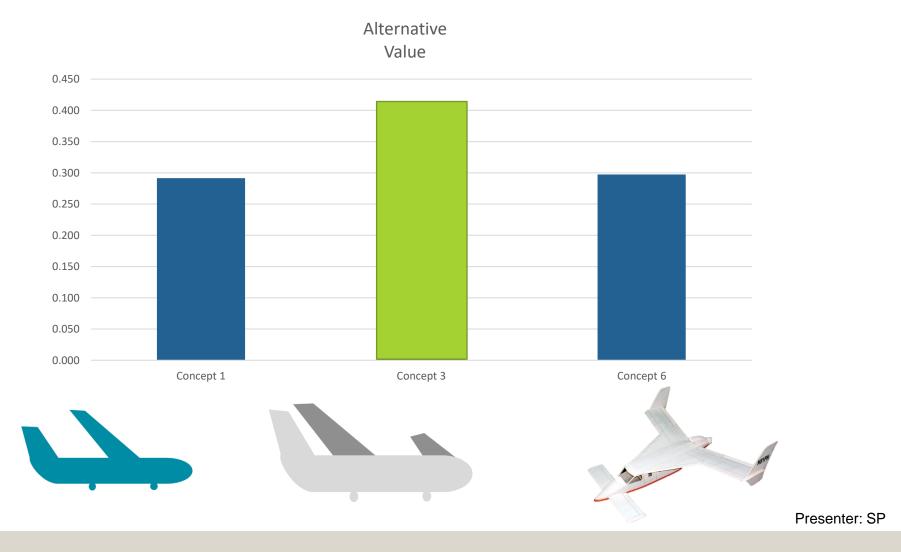


#### **Concept Comparison- AHP**





#### **Concept Comparison- AHP**



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# Concept Selection The Chosen One

Presenter – Adrian Moya



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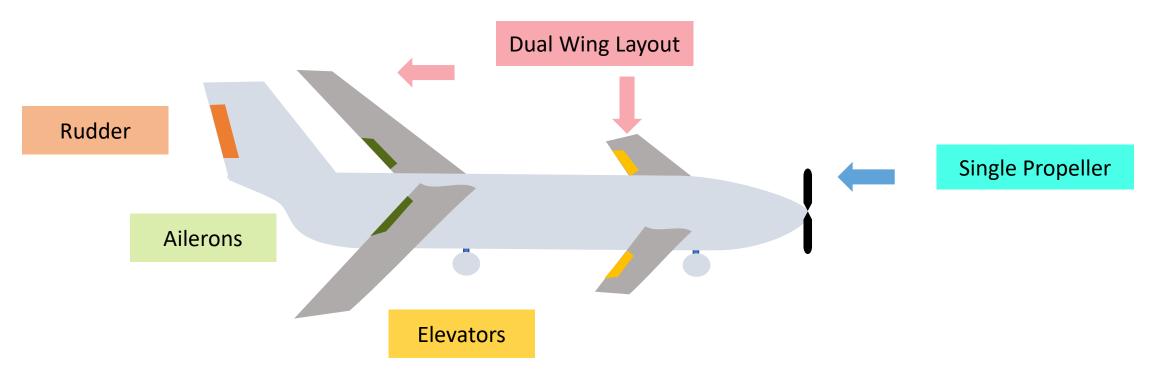
### Concept 3: Rutan Quickie Q2

Presenter: AM





#### Chosen Design



### Concept 3: Rutan Quickie Q2

Presenter: AM





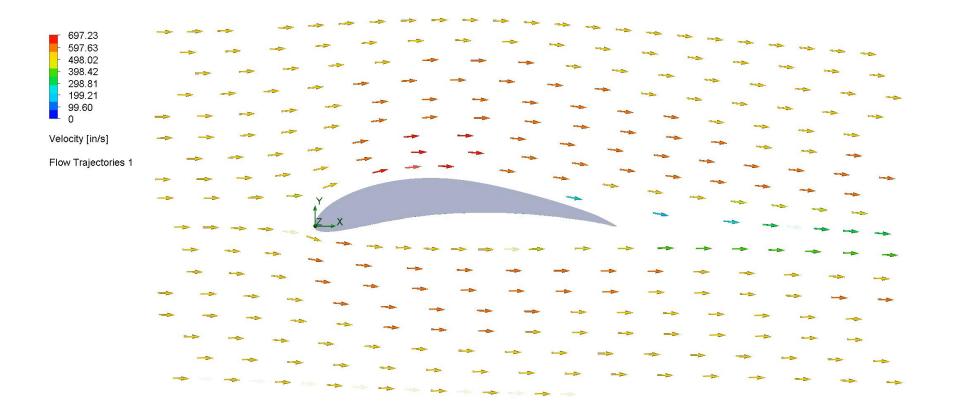
#### **Current and Future Work**

Presenter – Adrian Moya

FAMU-FSU 71

#### **Current Work – Fluid Analysis**

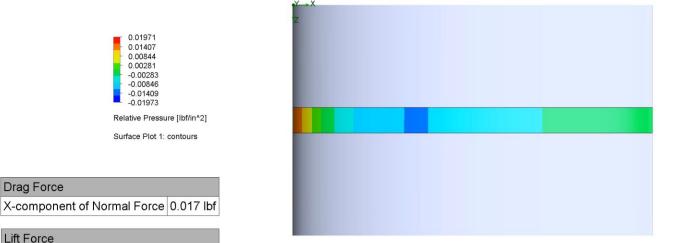
#### Eppler 423 Airfoil



Presenter: AM



### **Current Work – Fluid Analysis**



#### **Top Surface**

Drag Force

Lift Force

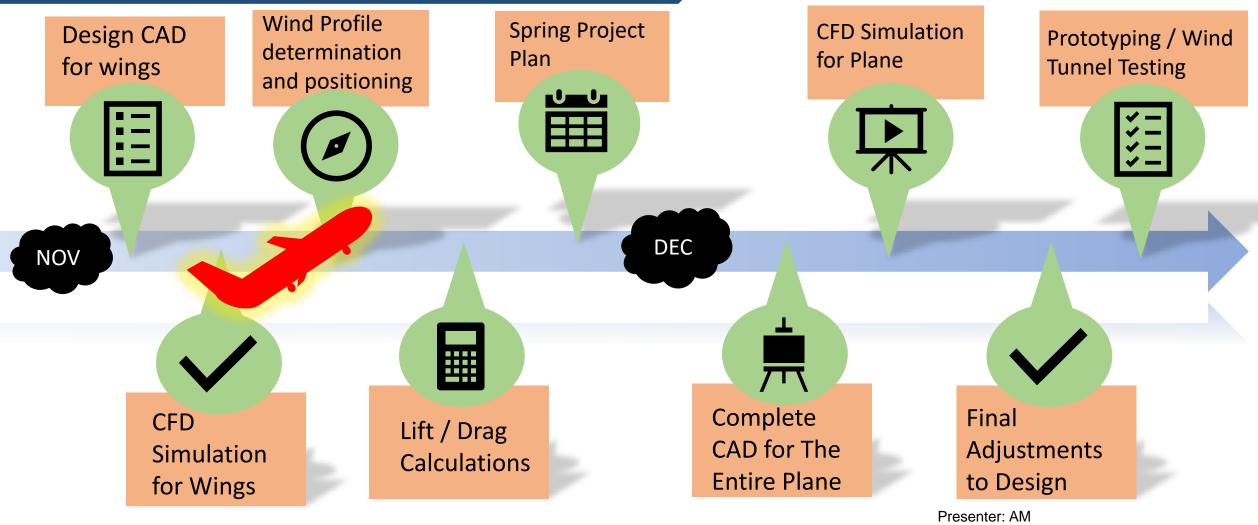
Y-component of Normal Force 0.036 lbf

**Bottom Surface** 

Presenter: AM



### **Project Timeline – FALL**





### References

SAE Aero Design Competition 2021 Rule Book. Available on: <a href="https://public.3.basecamp.com/p/38Lpy4uyTLpNkwTZbtwjgtBZ">https://public.3.basecamp.com/p/38Lpy4uyTLpNkwTZbtwjgtBZ</a>

Fundamentals of Aerodynamics. John D. Anderson Jr. 2011. 5<sup>th</sup> Edition. McGraw Hill Publications.

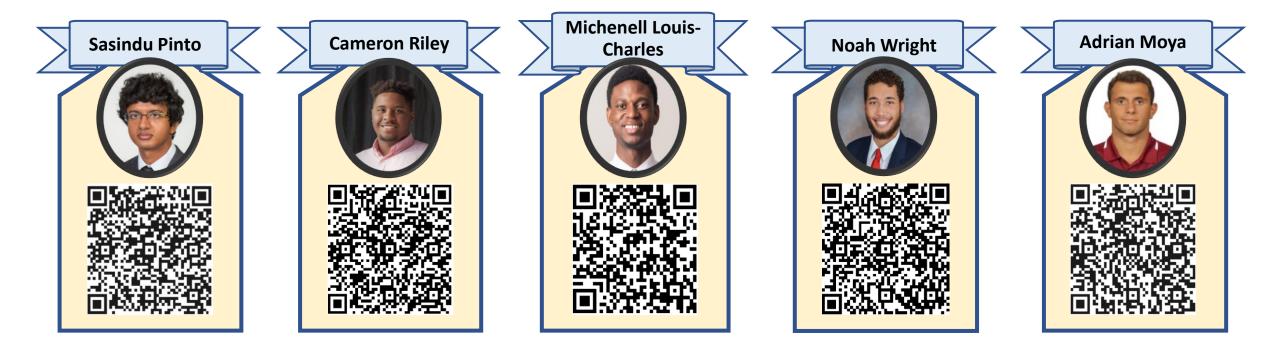
Fuselage Shapes. Academic. N.d. <u>https://enacademic.com/dic.nsf/enwiki/109692</u>

Tail Types. What-When-How. N.d. http://what-when-how.com/flight/tail-designs/

Presenter: AM



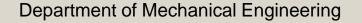
### Linked in Information



#### Criteria Comparison

Characteristic Comparison

**Final Rating Matrix** 





# Backup Slides

FAMU-FSU Engineering

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#### **Customer Needs**

#### Land within 400 ft

#### Takeoff within 100 ft

Presenter: CR



- Control Surface Movement
  - Controlling pitch ~ angle about x-axis
    - Servo Motor Torque produced ~ Greater than 66 oz-in
  - Controlling yaw ~ angle about y axis
    - Servo Motor Torque produced ~ Greater than 66 oz-in
  - Controlling roll ~ angle about z axis
    - Servo Motor Torque produced ~ Greater than 66 oz-in

Presenter: MLC

Roll



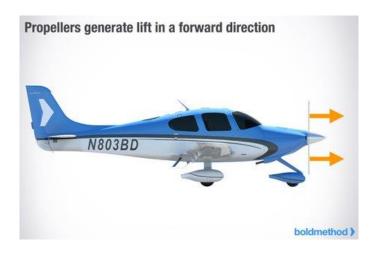
Longitudinal

l aberat

- Generating Drag (Air and Ground)
  - Coefficient of Drag ~ Greater than 1
  - Air Brake Force ~ 2-5 lbf
  - Coefficient of Rolling Friction ~ 0.03-0.06
  - Landing Velocity ~ Less than 25 mph
  - Landing Gear Force absorption ~ Greater than 55 lbs

Presenter: MLC

- Generate Thrust
  - Thrust Force ~ 15lbf
  - Propeller Diameter ~ 16in-20in
  - Electric Motor Power ~ 950W



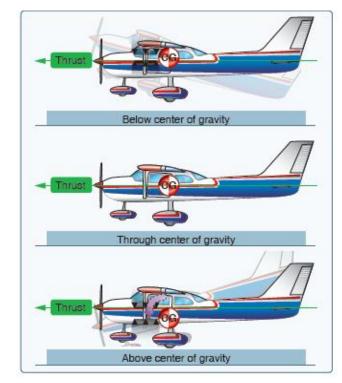


Figure 5-26. Thrust line affects longitudinal stability.

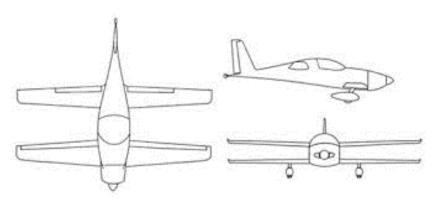
Presenter: MLC



- Deceleration
  - Coefficient of Drag ~ Greater than 1
  - Air Brake Force ~ 2-5 lbf
  - Coefficient of Rolling Friction ~ 0.03-0.06
  - Landing Velocity ~ Less than 25 mph
  - Landing Gear Force absorption ~ Greater than 55 lbs

### **Chosen Design**

• The winner of the concept selection was...



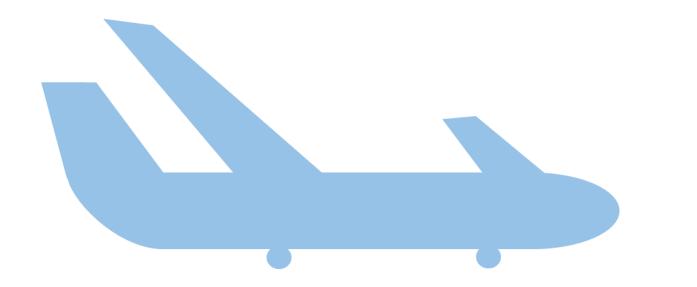
Rutan Quickie Q2!



### **Future Work**

- Pinpointing our wing position and profile
- CAD Modeling (SolidWorks)
- Fluid Analysis (SolidWorks, Ansys)
- Verifying the accuracy of SolidWorks CFD with simple wind tunnel tests

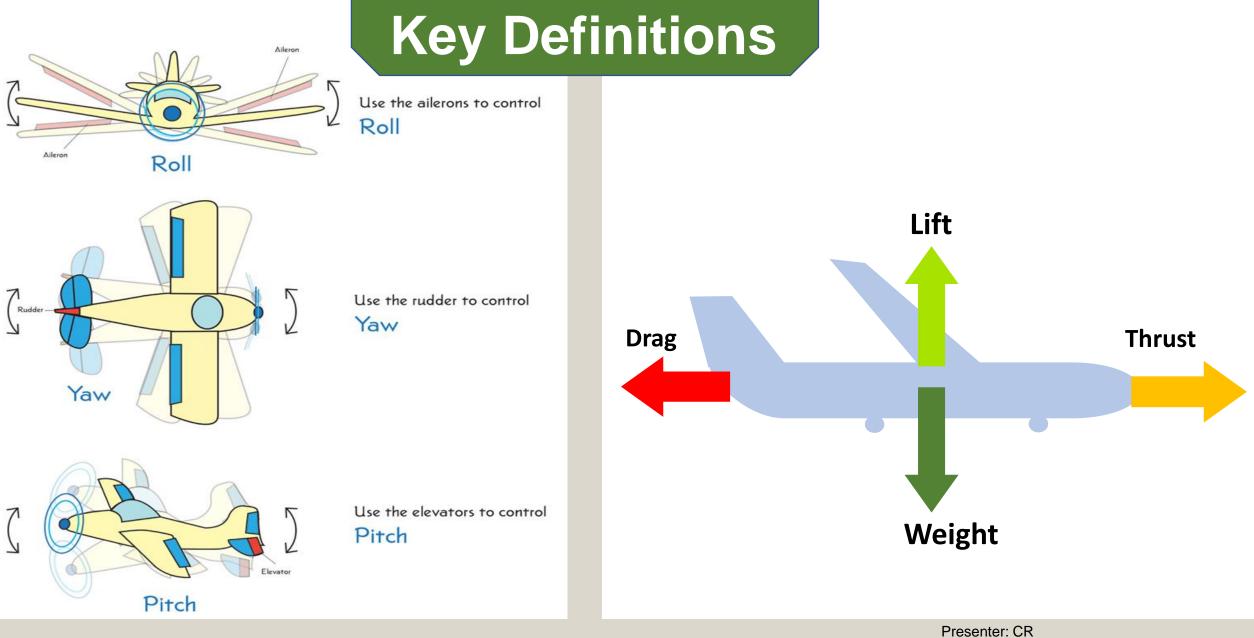
### **Customer Needs Considered**





**Department of Mechanical Engineering** 







- Control Surface Movement
  - Controlling pitch ~ angle about x-axis
    - Servo Motor Torque produced ~ Greater than 66 oz-in



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#### Control Surface Movement

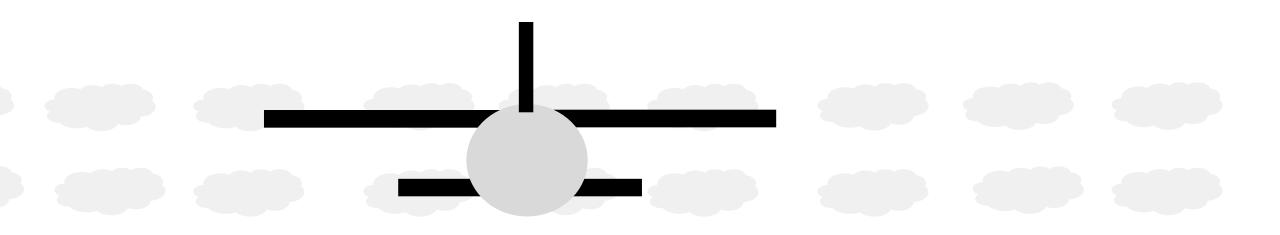
- Controlling yaw ~ angle about y axis
  - Servo Motor Torque produced ~ Greater than 66 oz-in





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- Control Surface Movement
  - Controlling roll ~ angle about z axis
    - Servo Motor Torque produced ~ Greater than 66 oz-in





# House of Quality



### **Binary Pairwise Comparison**

Binary Pairwise Comparison													
	1	2	3	4	5	6	7	8	9	10	11	12T	otal
1. Material	-	0	0	0	0	0	0	1	0	0	0	0	1
2. Stability	1	-	0	0	0	1	1	1	1	0	0	1	6
3. CG in front of CP	1	1	-	1	1	1	1	1	1	1	1	1	10
4. Meet takeoff/landing requirements	1	1	0	-	1	1	1	0	1	0	0	1	7
5. Wingspan meets restrictions	1	1	0	0	-	1	1	1	1	0	0	1	7
6. Sufficient Power	1	0	0	0	0	-	0	0	1	1	1	1	5
7. Maneuverability	1	0	0	0	0	1	-	0	1	0	0	1	4
8. Light Weight	0	0	0	1	0	1	1	-	1	1	0	1	6
9. Touch-down Impact	1	0	0	0	0	0	0	0	-	0	0	1	2
10. Ground Controls	1	1	0	1	1	0	1	0	1	-	1	1	7
11. Carry the Minimum Cargo Load Required	1	1	0	1	1	0	1	1	1	0	-	1	8
12. Easy to Load/Unload	1	0	0	0	0	0	0	0	0	0	0	-	1
Total	10	5	0	4	4	6	7	5	9	4	3	10	-



### House of Quality

	House of Quality												
		_		E	ngineering	Chara	cteristi	cs (***Fro	m Mạin Ta	rgets	***)		
Improvement Directi	ion				-	➡							
Units		lbf	lbf	lbf	degrees	ft/s	ft/s^2	degrees	seconds	lbs	ft/s^2	– psi	psi
Customer Requirements	lmportance Weight Factor	Lift	Drag	Thrust	Max Angle of Attack	Stall Speed	Acceleration	Control Surface Movement	Loading/ Unloading Time	Weight	Deceleration	Joint Strength	Material Strength
1. Material		1	1							9		9	g
2. Stability		5 9	3	3				9					
3. CG in front of CP	1	0 9	3	9	9	9		9		3			
4. Meet takeoff/landing													
requirements		7 9	3	9			9				9		
5. Wingspan meets													
restrictions		7 9	3		3	3		1				3	3
6. Sufficient Power		5 1	1	3			3	3		1	1		
7. Maneuverability		4			3	3		9		3		3	1
8. Light Weight		5 3		3			3			9	3		
9. Touch-down Impact		2						3		3	9	9	g
10. Ground Controls		7						1					
11. Carry the Minimum													
Cargo Load Required		3 9		3			3		9	9	3	9	9
12. Easy to Load/Unload		1							9	3		3	
Raw Score		365	96	228	123	123	120	215	81	191	128	135	124
Relative Weight %		18.92	4.98	11.82	6.38	6.38	6.22	11.15	4.20	9.90	6.64	7.00	6.43
Rank Order		1	11	2	6	6	10	3	12	4	8	5	g



# **Pugh Charts**



### Pugh Chart 1

Pugh Chart 1		Concepts							
			High			Μ	ediu	m	
Selection Criteria	2020 Competition Entry	1	2	3	4	5	6	7	8
Lift		+	+	+	-	-	+	-	-
Thrust		S	S	S	S	S	S	S	S
Control Surface	DATUM								
Movement	DATOW	+	+	+	+	S	+	S	S
Weight		-	S	-	-	-	S	-	S
Joint Strength		+	+	+	+	+	+	+	+
# of pluses		3	3	3	2	1	3	1	1
# of S's		1	2	1	1	2	2	2	3
# of Minuses		1	0	1	2	2	0	1	1

### Pugh Chart 2

Pugh Chart 2			Concepts			
			Hig	h	Medium	
Selection Criteria	Concept 2		1	3	6	
Lift		-		+	-	
Thrust		S		S	S	
Control Surface	Deture					
Movement	Datum	+		+	+	
Weight		-		-	-	
Joint Strength		S		S	S	
# of pluses			1	2	1	
# of S's			2	2	2	
# of Minuses			2	1	2	



# **AHP Criteria Comparison**





### **Criteria Comparison Matrix**

Development of a Candidate set of Criteria Weights {W}								
	Criteria Comparison Matrix							
	Lift	Thrust	Control Surface Movement		Weight	Joint Strength		
Lift	1.00	0.33		3.00	9.00	9.00		
Thrust	3.00	1.00		3.00	9.00	9.00		
Control Surface Movement	0.33	0.33		1.00	5.00	3.00		
Weight	0.11	0.11		0.20	1.00	0.11		
Joint Strength	0.11	0.11		0.33	9.00	1.00		
Sum	4.56	1.89		7.53	33.00	22.11		





### **Normalized Comparison Matrix**

	Normalized Criteria Comparison Matrix [NormC]								
Criteria Comparison Matrix									
	Lift	Thrust	Control Surface Movement	Weight	Joint Strength	Criteria Weight			
Lift	0.22	0.18	0.40	0.27	0.41	0.295			
Thrust	0.66	0.53	0.40	0.27	0.41	0.453			
Control Surface Movement	0.07	0.18	0.13	0.15	0.14	0.134			
Weight	0.02	0.06	0.03	0.03	0.01	0.029			
Joint Strength	0.02	0.06	0.04	0.27	0.05	0.089			
Sum	1.00	1.00	1.00	1.00	1.00	1.000			





### **Criteria Comparison Consistency Check**

Consistency Check								
{Ws}=[C]{W} Weighted Sum Vector	{W} Criteria Weights	Con={Ws}./{W} Consistency Vector						
1.911	0.490	3.899						
2.802	0.230	12.184						
0.796	0.140	5.683						
0.149	0.040	3.720						
0.478	0.100	4.780						

λ	CI	CR
Average	Consistency	Consistency
Consistency	Index	Ratio
6.053	0.027	0.051





### **AHP – Lift Tables**





### Lift Comparison Matrix

Lift Comparison								
	Concept 1	Concept 3	Concept 6					
Concept 1	1.00	0.33	3.00					
Concept 3	3.00	1.00	7.00					
Concept 6	0.33	0.14	1.00					
Sum	4.33	1.48	11.00					

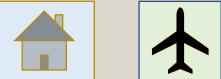




### **Normalized Lift Comparison Matrix**

Normalized Criteria Comparison Matrix [NormC]									
	Concept 1	Concept 2	Concept 6	Criteria Weight					
Concept 1	0.231	0.226	0.273	0.243					
Concept 2	0.692	0.677	0.636	0.669					
Concept 6	0.077	0.097	0.091	0.088					
Sum	1.000	1.000	1.000	1.000					
Mechanical Enginee	ring			FAN Eng					

Department of M

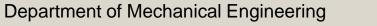




### Lift Consistency Check

Consistency Check 1								
{Ws}=[C]{W}		Con={Ws}./{W}						
Weighted Sum	{W} Criteria	Consistency						
Vector	Weights	Vector						
0.731	0.243	3.005						
2.015	0.669	3.014						
0.265	0.088	3.002						

λ	CI	CR
Average	Consistency	Consistency
Consistency	Index	Ratio
3.00703	0.00352	0.00676







### **AHP – Thrust Tables**





### **Thrust Comparison**

Thrust Comparison							
	Concept 1	Concept 3	Concept 6				
Concept 1	1.00	1.00		1.00			
Concept 3	1.00	1.00		1.00			
Concept 6	1.00	1.00		1.00			
Sum	3.00	3.00		3.00			





### **Normalized Thrust Comparison Matrix**

Normalized Criteria Comparison Matrix [NormC]					
	Concept 1	Concept 2	Concept 6	Criteria Weight	
Concept 1	0.333	0.333	0.333	0.333	
Concept 2	0.333	0.333	0.333	0.333	
Concept 6	0.333	0.333	0.333	0.333	
Sum	1.000	1.000	1.000	1.000	
Mechanical Enginee	ring			FAN Eng	





### **Thrust Consistency Check**

Consistency Check 2						
{Ws}=[C]{W} Weighted Sum	{W} Criteria	Con={Ws}./{W}				
Vector	Weights	Consistency Vector				
1.000	0.333	3.000				
1.000	0.333	3.000				
1.000	0.333	3.000				

λ	CI	CR
Average	Consistency	Consistency
Consistency	Index	Ratio
3.00000	0.00000	0.00000





## AHP – Control Surface Movement Tables

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### **Control Surface Comparison Matrix**

<b>Control Surface Movement Comparison</b>				
	Concept 1	Concept 3	Concept 6	
Concept 1	1.00	3.00	0.20	
Concept 3	0.33	1.00	0.20	
Concept 6	3.00	5.00	1.00	
Sum	4.33	9.00	1.40	





#### **Normalized Control Surface Comparison Matrix**

Normalized Criteria Comparison Matrix [NormC]				
	Concept 1	Concept 2	Concept 6	Criteria Weight
Concept 1	0.231	0.333	0.143	0.236
Concept 2	0.077	0.111	0.143	0.110
Concept 6	0.692	0.556	0.714	0.654
Sum	1.000	1.000	1.000	1.000
Aechanical Enginee	ring			FAN Eng

Department of M





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## **Control Surface Consistency Check**

Consistency Check 3			
•	{W} Criteria	Con={Ws}./{W} Consistency Vector	λ Avera Consis
0.697	0.236	2.959	2
0.320	0.110	2.898	
1.912	0.654	2.924	

CI	CR
Consistency	Consistency
Index	Ratio
-0.03642	-0.07004
	Consistency Index





# **AHP – Weight Tables**





## Weight Comparison Matrix

Weight Comparison				
	Concept 1	Concept 3	Concept 6	
Concept 1	1.00	0.33	3.00	
Concept 3	3.00	1.00	5.00	
Concept 6	0.33	0.20	1.00	
Sum	4.33	1.53	9.00	





## **Normalized Weight Comparison Matrix**

Normalized Criteria Comparison Matrix [NormC]				
	Concept 1	Concept 2	Concept 6	Criteria Weight
Concept 1	0.231	0.217	0.333	0.260
Concept 2	0.692	0.652	0.556	0.633
Concept 6	0.077	0.130	0.111	0.106
Sum	1.000	1.000	1.000	1.000
Mechanical Enginee	ring			FAN Eng

Department of M



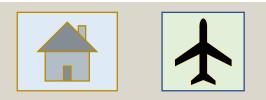


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## Weight Consistency Check

Consistency Check 4			
Ŭ	{W} Criteria	Con={Ws}./{W} Consistency Vector	
0.790	0.260	3.033	
1.946	0.633	3.072	
0.320	0.106	3.011	

λ	CI	CR
Average	Consistency	Consistency
Consistency	Index	Ratio
3.03871	0.01936	0.03723





# **AHP – Joint Strength Tables**

From Team 508





# Joint Strength Comparison Matrix (508)

Joint Strength Comparison				
	Concept 1	Concept 3	Concept 6	
Concept 1	1.00	1.00	1.00	
Concept 3	1.00	1.00	1.00	
Concept 6	1.00	1.00	1.00	
Sum	3.00	3.00	3.00	





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### **Normalized Joint Comparison Matrix** (508)

Normalized Criteria Comparison Matrix [NormC]				
	Concept 1	Concept 2	Concept 6	Criteria Weight
Concept 1	0.333	0.333	0.333	0.333
Concept 2	0.333	0.333	0.333	0.333
Concept 6	0.333	0.333	0.333	0.333
Sum	1.000	1.000	1.000	1.000
Mechanical Enginee	ering			FAN FAN

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#### Joint Strength Consistency Check(508)

Consistency Check 5			
{Ws}=[C]{W} Weighted Sum	Con={Ws}./{W} {W} Criteria Consistency		
Vector	Weights	Vector	
1.000	0.333	3.000	
1.000	0.333	3.000	
1.000	0.333	3.000	

λ	CI	CR
Average	Consistency	Consistency
Consistency	Index	Ratio
3.00000	0.00000	0.00000





# **Final Rating**



## **Final Rating Matrix**

Final Rating Matrix			
Selection			
Criteria	Concept 1	Concept 2	Concept 6
Lift	0.243	0.669	0.088
Thrust	0.333	0.333	0.333
Control			
Surface			
Movement	0.236	0.110	0.654
Weight	0.260	0.633	0.106
Joint Strength	0.333	0.333	0.333





