MICROGRAVITY MACHINE Design Review 6

Team 511





Propulsion Engineer



Recovery Engineer



Test & Systems Engineer



Engineer

John Tietsworth

Controls Engineer

Samuel Duval



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Sponsor and Advisor



FAMU-FSU College of Engineering



Florida Space Grant Consortium



Dr. Shayne McConomy

FAMU-FSU College of Engineering

Samuel Duval



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Project Objective

The objective of the project is to design a reproduceable system that can be dropped, achieve microgravity during its descent, and be safely recovered for reuse.



Samuel Duval

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Achieving Freefall



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Key Goals







Microgravity

Recoverable

Meet weight requirements

Reproduceable

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Competition Day







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Competition Day





Metric	Target
Time to Detect Disconnection	0.1s
Degrees of Freedom	1
Acceleration	9.81 $\frac{m}{s^2}$
Final Velocity	$5\frac{m}{c}$

$$\vec{a} = 9.81 \frac{m}{s^2}$$

Pedro Siman





Competition *** Day

Metric	Target
Time to Detect Disconnection	0.1s
Degrees of Freedom	1
Acceleration	9.81 $\frac{m}{s^2}$
Final Velocity	$5\frac{m}{s}$

Pedro Siman

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Competition *** Day



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Concept Generation



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Concept Generation



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Collin Gainer





Collin Gainer





Collin Gainer











Collin Gainer









Air is pulled in by the fan and expelled from nozzles ayload Moto

Collin Gainer





Collin Gainer







Collin Gainer

























Collin Gainer





Front View



Collin Gainer



Side View

5.5 ft (1.68 m)



Collin Gainer

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Side View



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Uncontrolled Simulation

1st portion: Vehicle falls under gravity with drag acting on it.

2nd portion: Vehicle slows after parachute is released

 $V \approx 4$ m/s 0.5 s after the parachute is opened

John Tietsworth



Uncontrolled Simulation

1st portion: Vehicle falls under gravity with drag acting on it.

2nd portion: Vehicle slows after parachute is released

 $V \approx 4$ m/s 0.5 s after the parachute is opened

100	-50	0 (m)	50	
100				
120				
140				
160				
光 ₁₈₀ -				
14 200 ·				
Ê 220				
240				
260				
280				
300				

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Controlled Simulation (PID)



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Chosen Microcontroller



FireBeetle ESP32

- 240 MHz
- Dual core processing

Compared to Arduino Mega

15x faster computation

Samuel Duval







Samuel Duval





Samuel Duval



Thrust Testing - BUILD







Samuel Duval



Thrust Testing - Adaptation



Samuel Duval



Max Drag	15.4 N	
Max Fan Thrust	28 N	
Duct Iteration	Maximum Trust	

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Max Drag	15.4 N	
Max Fan Thrust	28 N	
Duct Iteration	Maximum Trust	
1	7 N	

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Max Drag	15.4 N
Max Fan Thrust	28 N
Duct Iteration	Maximum Trust
1	7 N
2	8.2 N



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Max Drag	15.4 N	
Max Fan Thrust	28 N	
Duct Iteration	Maximum Trust	
1	7 N	
2	8.2 N	
3	8.6 N	



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Max Drag	15.4 N
Max Fan Thrust	28 N
Duct Iteration	Maximum Trust
1	7 N
2	8.2 N
3	8.6 N
4	11 N



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Max Drag	15.4 N
Max Fan Thrust	28 N
Duct Iteration	Maximum Trust
1	7 N
2	8.2 N
3	8.6 N
4	11 N
5	18.5 N



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PWM Signal vs. Thrust



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PWM Signal vs. Thrust



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Servo Testing Results





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Testing Plans





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Parachute Testing Results





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Payload Testing Results





Bearings and Payload Caps Aligned

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Payload Testing Results







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Payload Testing Results









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Testing Plans

- 1. Thrust curve for the EDF
- 2. Servo stall torque
- 3. Parachute release
- 4. Payload movement
- 5. Control system



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Future Work



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Double check test conditions before running test

Acquire parts ASAP

Double check tolerances before printing

Expect setbacks

Test parts when first acquired

Research parts before purchasing

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References

Images:

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