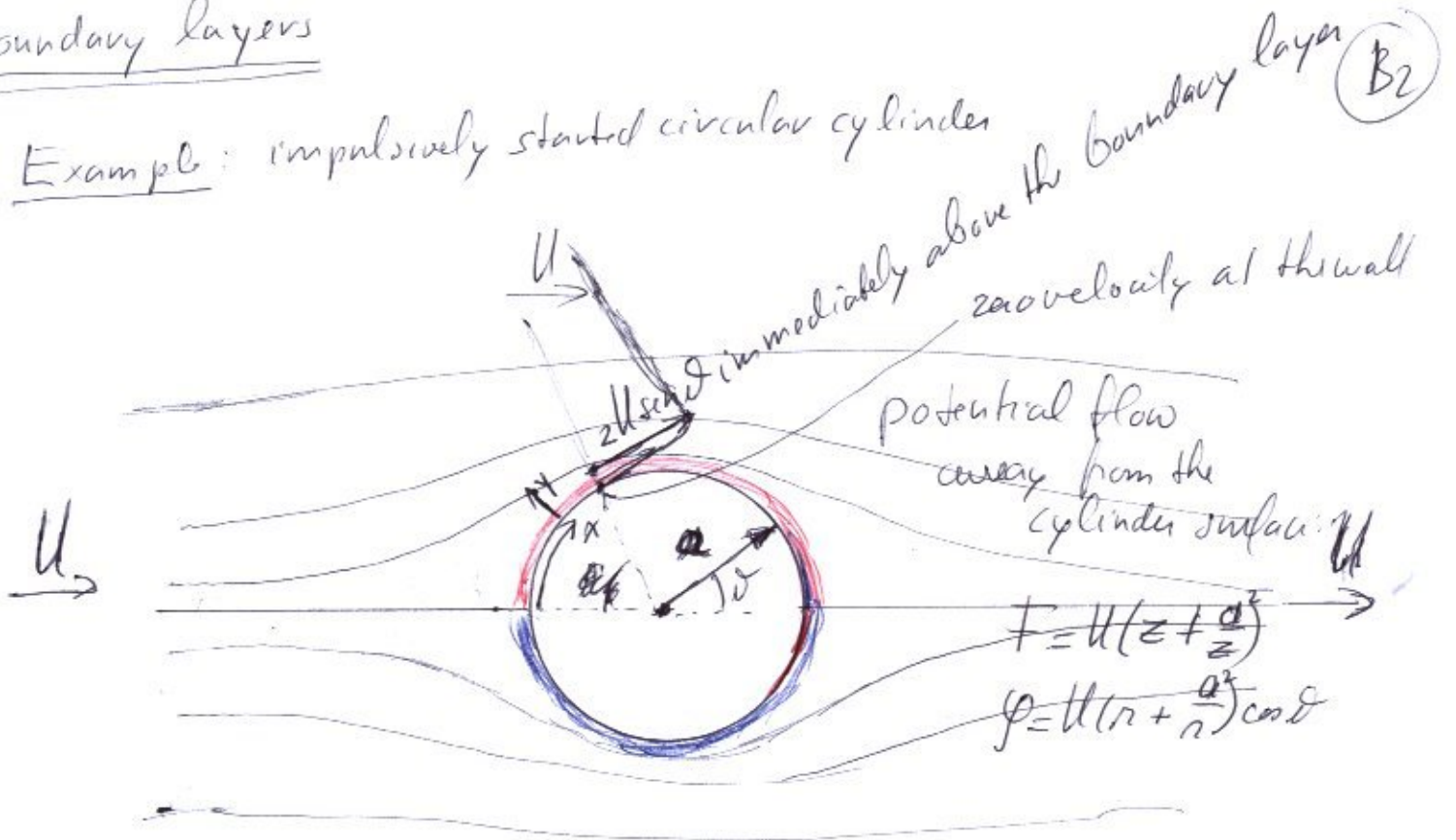


Boundary layers

Example: impulsively started circular cylinder



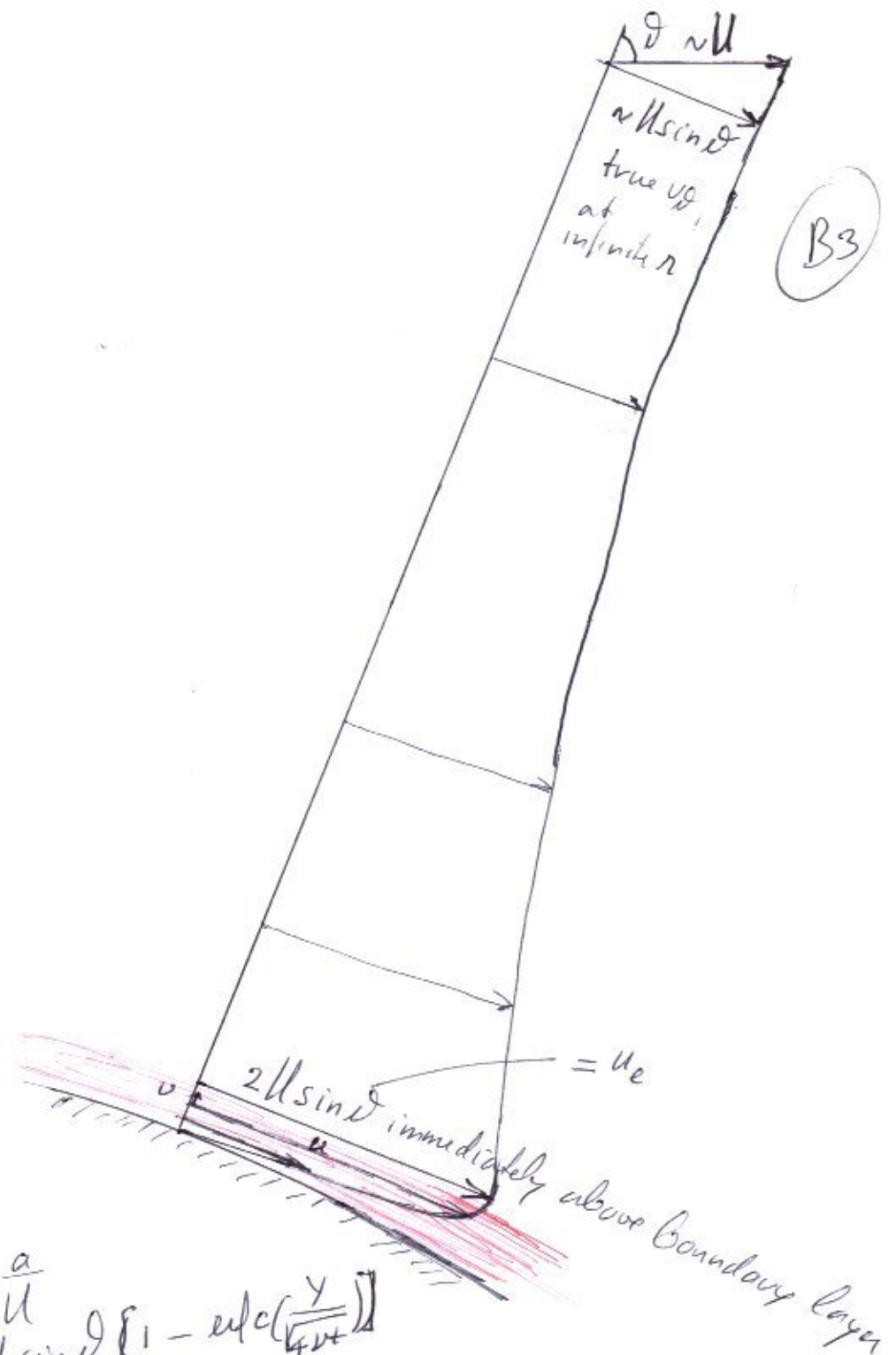
For $t \ll \frac{1}{3} \frac{a}{U}$: Stokes layer $\frac{U}{\nu}$

For $t < \sim \frac{1}{3} \frac{a}{U}$ ($\frac{1}{3}$ radius thick); no reversed velocity

For $t > \sim \frac{1}{3} \frac{a}{U}$ but $t < \sim 1.5 \frac{a}{U}$: thin layer of reversed velocity near the rear surface.

For $t \approx 1.5 \frac{a}{U}$: Van Dommelen & Shen singularity.

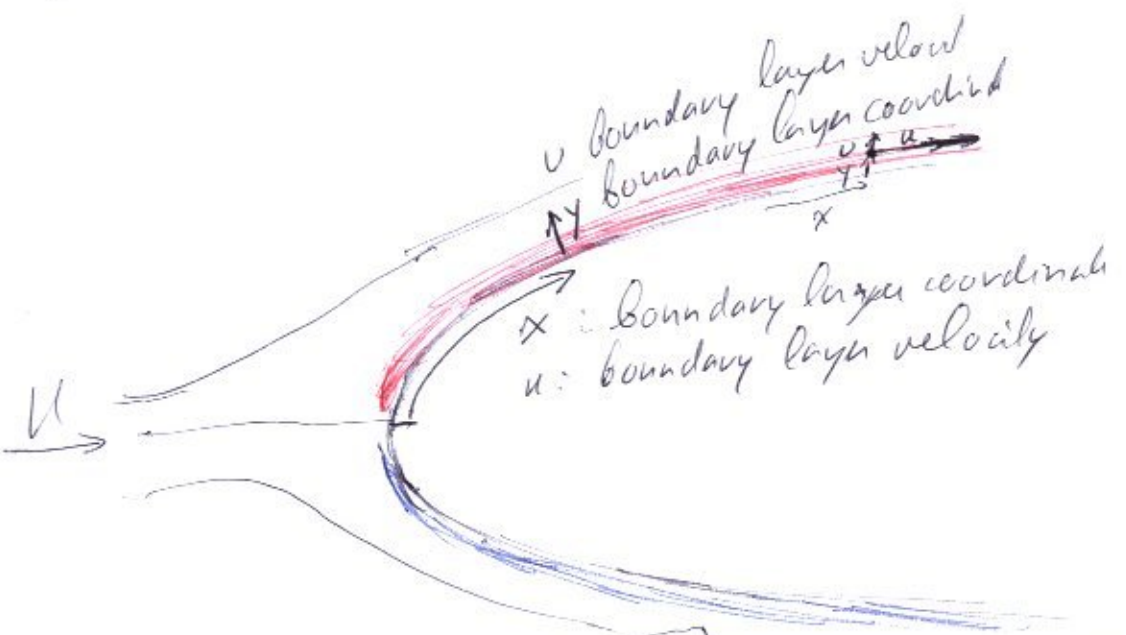




For $t \ll \frac{1}{3} \frac{a}{U}$
 $u \sim 2U \sin \theta \left[1 - \exp\left(-\frac{y}{4\sqrt{ut}}\right) \right]$

Boundary Layer Coordinates
Example: parabola (no separation)

B1



Boundary layer coordinates
 x : arc length along the wall
 y : distance from the wall
 u : velocity component parallel to the wall
 v : velocity component normal to the wall

NOT Cartesian coordinates
(except for a flat plate)

Boundary layer equations

Apply to the limit $Re \rightarrow \infty$ $Re = \frac{U \ell}{\nu}$ \rightarrow typical length.

Simpler approach: look at limit $\nu \rightarrow 0$

Since the x and y coordinates look almost Cartesian on scales comparable to the boundary layer thickness, the equations look Cartesian:

$$\text{Continuity: } \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$$

$$x\text{-momentum: } \frac{\partial u}{\partial t} + u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} = -\frac{1}{\rho} \frac{\partial p}{\partial x} + \nu \left(\frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} \right)$$

$$y\text{-momentum: } \frac{\partial v}{\partial t} + u \frac{\partial v}{\partial x} + v \frac{\partial v}{\partial y} = -\frac{1}{\rho} \frac{\partial p}{\partial y} + \nu \left(\frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 v}{\partial y^2} \right)$$

But some terms drop out when $\nu \rightarrow 0$. Outside the boundary layer, that are the viscous terms. But inside the boundary layer, things change:

here $u = \tilde{u} \left(x, \frac{y}{\sqrt{\nu}} \right)$ where \tilde{u} is a well behaved function of its arguments

$$v = \sqrt{\nu} \tilde{v} \left(x, \frac{y}{\sqrt{\nu}} \right)$$

$$p = \tilde{p} \left(x, \frac{y}{\sqrt{\nu}} \right)$$

where $\tilde{u}, \tilde{v}, \tilde{p}$ are well behaved functions of their arguments (finite derivatives)

$$\text{so } \frac{\partial u}{\partial y} = \frac{\partial \tilde{u}}{\partial (y/\sqrt{\nu})} \frac{\partial (y/\sqrt{\nu})}{\partial y} = \frac{\partial \tilde{u}}{\partial (y/\sqrt{\nu})} \frac{1}{\sqrt{\nu}} \quad \frac{\partial^2 u}{\partial y^2} = \frac{\partial^2 \tilde{u}}{\partial (y/\sqrt{\nu})^2} \frac{1}{\nu} \rightarrow \infty \text{ when } \nu \rightarrow 0$$

so $\nu \frac{\partial^2 u}{\partial y^2}$ does not go to zero even if ν does.

(The derivatives of u w.r.t. y are large)

But $\frac{\partial^2 u}{\partial x^2}$ is not large, so $\nu \frac{\partial^2 u}{\partial x^2}$ does still drop out. B5
 become negligibly small

$$\frac{\partial u}{\partial x} \text{ finite and } \frac{\partial v}{\partial y} = \frac{\sqrt{\nu} \frac{\partial \tilde{v}}{\partial y \sqrt{\nu}}}{\sqrt{\nu}} \underline{\text{finite too}}$$

Boundary layer equations (2D incompressible)

continuity: $\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$

x-momentum: $\frac{\partial u}{\partial t} + u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} = -\frac{1}{\rho} p_x + \nu \frac{\partial^2 u}{\partial y^2}$

y-momentum: $0 = -\frac{1}{\rho} p_y$

Unfortunately non-linear

~~Can integrate y-momentum immediately~~
 ~~$p = p(x, t) = p(x, t)$ immediately above the boundary layer~~
 ~~$= p_e - \frac{1}{2} \rho \vec{v}^2$~~
 ~~$\approx p_{\text{at flow, wall}}(x, t)$~~

Can integrate y-momentum equation immediately

$$p = p(x, t)$$

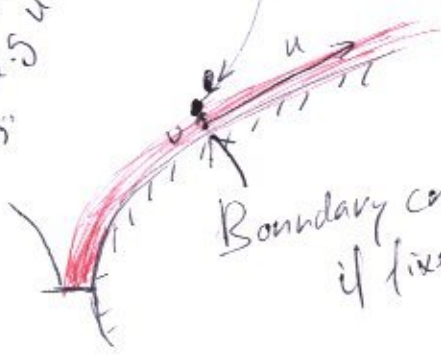
Let e (external) denote a point just above the boundary layer. Then p inside the boundary layer is

$$p = p_e(x, t) \stackrel{\text{Bernoulli}}{\approx} p_e - \frac{1}{2} \rho \vec{v}_e^2 \approx p_e - \frac{1}{2} \rho \vec{v}_{\text{at flow at wall}}^2$$

Eg, $p = p_e - \frac{1}{2} \rho (2U \sin \theta)^2$ for impulsively started circular cylinder

Boundary and initial conditions

Initial conditions
at $x=0$
e.g. $u=0$ by symmetry



boundary condition at edge of boundary layer
 $u \rightarrow u_e$ when $\frac{y}{\delta} \rightarrow \infty$

where u_e is the potential flow velocity for $y \rightarrow \infty$

Boundary condition at $y=0$:
if fixed solid wall: $u=v=0$